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High Temperature Coating Study to Reduce Contact Stress Damage of Ceramics

**Final Report** 

J. Schienle and J. Smyth

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J. Schienle and J. Smyth

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### CERAMIC TECHNOLOGY FOR ADVANCED HEAT ENGINES PROJECT

### J. Schienle, J. Smyth

### ABSTRACT

Silicon nitride ( $\mathrm{Si}_3N_4$ ) and silicon carbide (SiC) ceramics are susceptible to contact stress damage at ceramic component interfaces in gas turbine engines. The objective of this research program was to develop adherent coating compositions and procedures that will improve contact-stress damage resistance of  $\mathrm{Si}_3N_4$  and SiC.

Yttria (Y2O3) stabilized zirconia (ZrO2) coatings were applied by electron beam - physical vapor deposition (EB-PVD) to three substrate materials: reaction bonded  $Si_3N_4$  (RBSN), sintered  $Si_3N_4$ , and sintered SiC (SSC). Several controlled substrate pretreatments (both mechanical and chemical) were used to aid in obtaining a better understanding of the mechanisms of coating adherence. Each coating/substrate system was evaluated for coating adherence before and after oxidation exposure. The coating/substrate interfaces were evaluated for microstructural characteristics. Based on the results, pretreatment and coating approaches projected to improve coating adherence were selected. Specimens were prepared to evaluate the coating adherence before and after oxidation exposure. The systems exhibiting the best coating adherence after oxidation exposure were identified. Additional specimens for each of these systems were prepared for analysis of the coating thermal spall resistance, friction, and contact damage resistance. The friction and contact damage resistance of uncoated RBSN, SSN, and SSC were also evaluated.

### 1.0 INTRODUCTION

Significant improvements in turbine engine operating efficiencies can be obtained through the use of uncooled ceramic components at material temperatures above those attainable with high-temperature metal alloys. Both The Department of Energy (DOE) and The Department of Defense (DOD) advanced turbine engine programs provide evidence that improved performance can be realized. However, these programs have also demonstrated that an expanded technology base is required to achieve reliable, cost-effective ceramic components.

During the past decade, a number of programs have focused on introducing ceramic materials, such as silicon nitride ( $\mathrm{Si}_3\mathrm{N}_4$ ) and silicon carbide ( $\mathrm{Si}_3\mathrm{C}$ ), into gas turbine engines. Due to the hard, brittle nature of ceramic materials, high localized surface stresses in contact regions do not redistribute as in metals. In many cases,

the localized stresses exceed the baseline strength of the ceramic, thus damaging the surface of a component and reducing its strength. As a result, contact stress damage can cause ceramic turbine engine components to fail unpredictably and prematurely.

Finite-element analyses of ceramic contacts show that localized tensile stresses can occur under certain contact conditions. The analytically derived stress distribution is shown schematically in Figure 1. For uniaxial loading, tensile stresses are generated on the surface at the edges of the contact zone, but the magnitude is small compared to the compressive stress beneath the surface. However, when a sliding force is added (i.e., a bi-axial loading condition), a large tensile spike occurs on the surface at the trailing edge of the contact zone. Four of the factors controlling the magnitude of this tensile stress spike are

- o Contact load
- o Contact geometry
- Coefficient of friction
- o Elastic moduli

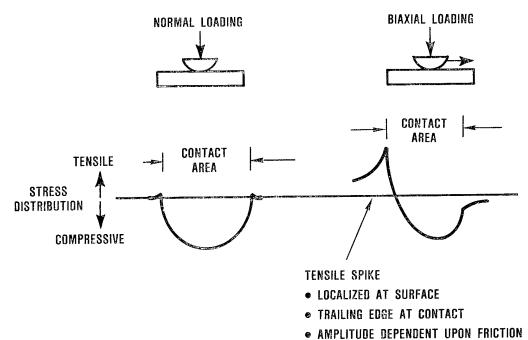


Figure 1. Biaxial loading generates a large tensile stress spike.

Contact load and contact geometry have a very significant effect on contact stresses and can be partially controlled through proper engine and component design, as well as through the control of dimensional tolerances of mating surfaces. In addition, significant benefits can be achieved if the sensitivity of ceramic components to contact stress damage can also be lessened by reducing the relative coefficient of friction (to reduce the sliding force component) and

the elastic moduli (to distribute the load over a larger area). A primary approach to controlling these factors is the use of coatings.

The feasibility of using ceramic coatings to improve the contact stress damage resistance of  $\mathrm{Si}_3\mathrm{N}_4$  and  $\mathrm{SiC}$  ceramics has been demonstrated using plasma sprayed oxide coatings. However, the coating adherence was not adequate for long term use in gas turbine environments. The objective of this research program was to develop coating compositions and procedures that will yield long term adherence and reduce or eliminate contact-stress damage to  $\mathrm{Si}_3\mathrm{N}_4$  and  $\mathrm{SiC}$  ceramics.

This program used an alternate coating method, electron beam physical vapor deposition (EB-PVD), as the coating process because it offers good control of coating composition, thickness, and morphology. The coating composition selected for evaluation was 20 percent yttria ( $Y_2O_3$ ) stabilized zirconia ( $ZrO_2$ ). The Garrett Turbine Engine Company (GTEC) uses this EB-PVD coating composition as a thermal barrier coating (TBC) for metallic components. Also, the feasibility of using this coating on  $Si_3N_4$  and SiC ceramics was demonstrated prior to the start of the program.

Three substrate materials were selected for this study: reaction bonded  $Si_3N_4$  (RBSN), sintered  $Si_3N_4$  (SSN), and sintered SiC (SSC).

The program was divided into four technical tasks:

- o Task 1 Coating Adherence and Characteristics Investigation
- o Task 2 Advanced Pretreatment and Coating Studies
- o Task 3 Contact Stress Testing and Friction Measurements
- o Task 4 Post-Contact Strength Measurements

### 2.0 SUMMARY

The objective of this research program was to develop adherent coating compositions and procedures that will reduce or eliminate contact-stress damage to  $Si_3N_4$  and SiC heat engine components. The program was divided into four technical tasks:

- o Task 1 Coating Adherence and Characteristics Investigation
- o Task 2 Advanced Pretreatment and Coating Studies
- o Task 3 Contact Stress Testing and Friction Measurements
- o Task 4 Post-Contact Strength Measurements

This four task approach addressed the application, adherence, and contact load tolerance of yttria stabilized zirconia applied by electron beam - physical vapor deposition (EB-PVD). The substrate materials evaluated were reaction bonded Si<sub>3</sub>N<sub>4</sub> (RBSN), sintered Si<sub>3</sub>N<sub>4</sub> (SSN), and sintered SiC (SSC). During Task 1, coating adherence evaluations were conducted to obtain an increased understanding of the coating adherence mechanisms. Several substrate pretreatments were evaluated for their effect on as-coated adherence and adherence after static air The substrate pretreatments included diamond grinding oxidation. (baseline), lapping, oxidation, etching, and sputtering interlayers of Al, Si, Mo, or Ti. The Task 1 adherence information was then used in Task 2 to develop coating/substrate systems with improved adherence characteristics. The Task 2 pretreatment approaches focused on preventing interfacial oxidation and improving the mechanical adherence. The former approach included the application of oxygen diffusion barrier interlayers of Al<sub>2</sub>O<sub>3</sub>, AlN, or mullite; the latter included roughening the substrate surface before coating. Other pretreatments evaluated included high purity interlayers (Chemical Vapor Deposition (CVD) SiC) and ion mixing. The adherence of ZrO2 deposited at higher temperatures and EB-PVD mullite coatings were also evaluated. Task 2 systems exhibiting the best coating adherence were then evaluated for contact damage resistance in Tasks 3 and 4,

### TASK 1

Coating studies conducted under Task 1 demonstrated that the EB-PVD process could produce adherent coatings of  $Y_2O_3$  stabilized  $ZrO_2$  on machined surfaces of RBSN, SSN, and SSC (Baseline). Analyses of coating/substrate interfaces suggested the presence of a thin layer of zirconium silicate at the interface. Neither oxidation or etching pre-treatments of the machined substrates changed the level of coating adherence relative to the baseline. All of the sputtered interlayers produced less adherent coatings relative to the baseline.

A major conclusion of Task 1 was that mechanical mechanisms contribute significantly to overall coating adherence. RBSN, which has the roughest surface of the materials evaluated due to its 15 percent residual porosity, consistently exhibited better adherence than SSN

and SSC substrates irrespective of substrate pretreatment. Also, lapping the machined substrates resulted in decreased adherence relative to the baseline. The effect of lapping was more significant for SSN and SSC than for RBSN. This difference was not surprising since lapping should not change the frequency and distribution of the RBSN's surface porosity. These results support the predominance of mechanical adherence for RBSN.

Static air oxidation of coated specimens significantly degraded the coating adherence for all coating/substrate systems. Analyses of coating/substrate interfaces indicated that the adherence loss was caused by additional growth of the zirconium silicate bond layer. The adherence loss observed for the sputtered interlayer specimens was probably due to oxidation of the metallic interlayer. Since  $\text{ZrO}_2$  is known to be permeable to oxygen, oxidation of the  $\text{Si}_3\text{N}_4$  and SiC substrates was anticipated. Although the oxidation and interlayer pretreatments were aimed at inhibiting degradation in interfacial bonding, these pretreatments were unsuccessful in preventing adherence loss. It was concluded that approaches aimed at reducing or eliminating the oxidation problem should be emphasized in subsequent tasks.

### TASK 2

Coating studies under Task 2 demonstrated the feasibility of using an oxygen diffusion barrier between the EB-PVD  $\rm ZrO_2$  coating and the substrate to prevent adherence loss on oxidation. CVD  $\rm Al_2O_3$  interlayers on RBSN yielded excellent as-coated adherence which was retained after 100 hours at 1200C in static air. CVD AlN interlayers on RBSN resulted in relatively poor initial adherence but excellent adherence after oxidation. The AlN evidently oxidized to form  $\rm Al_2O_3$  which improved adherence. A sol-gel  $\rm Al_2O_3$  interlayer worked best for SSN and SSC yielding fair adherence both before and after oxidation.

Task 2 studies also demonstrated that  $ZrO_2$  adherence could be improved significantly by roughening the substrate surface. Laser-texturing improved both the as-coated and oxidized coating adherence for SSC. This was particularly encouraging since no interlayer was used to prevent oxidation at the interface. Results were not as encouraging for laser textured RBSN and SSN. However, the surface cavities were much deeper than on SSC and were not successfully filled in with  $ZrO_2$  which suggests potential for improvement. One negative aspect of laser texturing is that improved coating adherence is accompanied by a strength loss associated with the surface cavities.

The other Task 2 pretreatment and coating approaches did not provide improved coating adherence. High purity CVD SiC interlayer and ion mixing pretreatments decreased coating adherence. Depositing EB-PVD ZrO<sub>2</sub> at a higher temperature (1040C rather than 980C) did not have any effect on coating adherence and attempts to deposit mullite coatings by EB-PVD were unsuccessful (the silica in the mullite evaporated more easily than the alumina resulting in silica rich coatings).

The two pretreatments yielding the best  ${\rm ZrO}_2$  coating adherence for each substrate were selected for testing in Tasks 3 and 4. Emphasizing the coating adherence after oxidation, the pretreatments shown in Table 1 were selected.

Table 1. Pretreatments selected for tasks 3 and 4

RBSN	SSN	SSC
CVD Al <sub>2</sub> O <sub>3</sub>	sol-gel Al <sub>2</sub> 0 <sub>3</sub>	sol-gel Al <sub>2</sub> O <sub>3</sub>
CVD AlN	laser machining	laser machining

### TASK 3

During Task 3, each coating/substrate system was evaluated for strength, thermal spalling resistance, and friction.

The strength characteristics were determined by comparing the post-oxidation (1200C for 100 hours) flexure strength of each coating/substrate system with the as-machined substrates. Laser-textured substrates were also evaluated since Task 2 results identified a strength loss associated with this process.

Both coated RBSN systems exhibited residual effects on strength. A strength loss was observed at room temperature which was recovered at elevated temperatures. At 1375C, both systems exhibited higher strengths than the baseline.

Both coated SSN systems exhibited a strength loss. However, the strength loss was due to the degradation of the SSN substrate during exidation rather than to the pretreatment and/or coating procedures.

Both coated SSC system exhibited no significant changes in strength relative to as-machined SSC.

The thermal spalling resistance of each coating/substrate system was evaluated using cyclic thermal exposures to 1000C and 1375C. The results were encouraging considering the large thermal expansion mismatch between the coating and the substrate. ZrO2/CVD Al2O3/RBSN had the best spalling resistance (no spalling observed). ZrO2/laser-textured SSN and ZrO2/laser-textured SSC performed nearly as well, exhibiting only minor edge spalling in some cases during 1000C testing. ZrO2/CVD AlN/RBSN exhibited relatively good spalling resistance; only one of four spalled during 1375C tests. ZrO2/sol-gel Al2O3/SSN and ZrO2/sol-gel Al2O3/SSC had poor spalling resistance. Limited spalling occurred during 1000C cyclic exposures, but frequent spalling occurred during the 1375C tests.

The friction characteristics of the substrate materials and each of the ZrO<sub>2</sub> coated systems were determined up to 1375C. Contact test procedures were used for friction assessment so the same specimens could be flexure tested for retained strength to assess contact damage resistance. First, the thresholds for contact damaging uncoated substrates were determined using a point contact geometry. Then, the friction characteristics of the coated systems were determined using

loading conditions below and above the thresholds for damaging the uncoated substrate (the coated specimens tested above the substrate damage thresholds were used to assess contact damage resistance under Task 4). When testing above the damage thresholds, frequent spalling was observed. The results suggested the coatings would not provide contact damage resistance under cyclic loading conditions.

Surprisingly, the ZrO<sub>2</sub> coatings had higher friction coefficients than the uncoated substrates, and the coefficients of friction increased with temperature. The higher friction coefficients relative to the uncoated substrate may have resulted from the SSC contact pin becoming embedded in the low modulus coating, which may create a plowing effect and increase apparent friction. A potential solution is to deposit a dense layer over the porous, compliant coating to inhibit the plowing effect. In addition, the increased friction at elevated temperature is due to the oxidation of the SSC contact pin. Therefore, the increased friction at elevated temperature would not be expected if the contact pins were also coated since two stable oxides would be in contact.

### TASK 4

During Task 4, post-contact flexure test results demonstrated that the coatings can prevent contact damage.

No contact damage resulted from room temperature contact testing.

At 1000C contact damage was prevented in most cases. A few specimens did exhibited contact damage. In these few cases the coating remained adherent throughout testing. It appears that well bonded EP-PVD ZrO<sub>2</sub> coatings can transmit damage to the substrate under certain conditions.

During 1200C and 1375C contact testing, contact damage was prevented for all systems except for the ZrO<sub>2</sub>/sol-gel Al<sub>2</sub>O<sub>3</sub>/SSN system. All specimens from this system spalled at 1200C and exhibited contact damage.

In conclusion, a technology base for depositing adherent EB-PVD  $Y_2O_3$  stabilized  $ZrO_2$  coatings on  $Si_3N_4$  and SiC was developed. These EB-PVD coatings improved the contact damage resistance of  $Si_3N_4$  and SiC. However, uncoated  $Si_3N_4$  and SiC substrates also had much higher contact damage resistance than originally anticipated. Because of the high load conditions required to damage the uncoated substrate, coated specimen tested under the same severe conditions tended to exhibit coating spalling during the single cycle contact tests. Therefore, additional work is required to obtain coatings with cyclic durability. The coating technology developed under this program may have immediate applicability for impact resistance coatings or non-stick coatings in heat engines.

### 3.0 TECHNICAL PROGRESS SUMMARY

### 3.1 TASK 1 - COATING ADHERENCE AND CHARACTERISTICS INVESTIGATION

Task 1 focused on determining the chemical and mechanical mechanisms of coating adherence. Based on Task 1 results, coating approaches projected to improve the adherence of the EB-PVD zirconia were selected for Task 2, Advanced Pretreatment and Coating Studies.

### 3.1.1 Specimen Preparation and Inspection

The substrate materials used in this program were reaction bonded silicon nitride (RBSN)\*, sintered silicon nitride (SSN)\*\*, and sintered silicon carbide (SSC)\*\*\*. RBSN and SSC were selected because substantial baseline strength and contact data were available. SSN was selected to provide a dense Si<sub>3</sub>N<sub>4</sub> to compare with the porous RBSN. The substrate test bars for Task 1 were selected from existing stock at GTEC and were diamond ground with a 320-grit wheel. The test bars were 0.250 by 0.125 inches in cross section and at least 1.0 inch in length. Half of these specimens were lapped smooth on one surface. The variation in surface roughness between the diamond ground and lapped surfaces provided for an assessment of the contribution of mechanical factors to the overall adherence.

Each specimen received one of the nine pretreatments listed in Table 2.

Surface Treatments	As-machined	Lapped
No Treatment	6	6
Light Oxidation <sup>+</sup>	6	6
Heavy Oxidation <sup>+</sup>	6	6
Sputter Etch++	6	6
Reactive Ion Etch++	6	6
Al Interlayer++	6	6
Si Interlayer <sup>++</sup>	6	6
Mo Interlayer++	6	6
Ti Interlayer++	6	6

Table 2. Task 1 precoating surface treatments

<sup>108</sup> specimens each of RBSN, SSN, and SSC (Total of 324 specimens)

<sup>&</sup>lt;sup>+</sup>Pretreatment conducted at GTEC

<sup>++</sup>Pretreatment conducted at Temescal, Berkeley, CA

<sup>\*</sup>RBN-104 from the AiResearch Casting Company (ACC) Torrance, California.

<sup>\*\*</sup>Code 2 from ACC.

<sup>\*\*\*</sup>Hexoloy SA from Sohio Engineered Materials, Niagara Falls, New York.

Specimens with no surface treatment were used as a baseline for comparison. Preliminary studies suggested that the EB-PVD zirconia coatings adhere relatively well to substrates with no pretreatment.

Light and heavy oxidation pretreatments were conducted to study the effect of growing a silica (SiO<sub>2</sub>) layer on the substrate prior to zirconia coating. Since zirconia is permeable to oxygen, it was anticipated that the substrate would oxidize during post-coating heat treatment. The light and heavy oxidation pretreatments were static air exposures at 1000C for 1 hour and 1204C for 4 hours, respectively.

Sputter etching was conducted to study the effect of etching all silica off the substrate prior to coating. Due to heat generated during machining and lapping, minor oxidation of the substrate can occur. Each substrate was RF sputter etched with argon ions.

Reactive ion etching was conducted to determine the effect of removing material from the substrate surface. The reactive etching agent used was silicon tetrafluoride ( $SiF_4$ ).

Four sputter deposited coatings, nominally one micron in thickness, were used as interlayers between the zirconia coating the substrate: aluminum (Al), silicon (Si), molybdenum (Mo), and titanium (Ti). Al was deposited anticipating that it would oxidize during heating to form alumina (Al $_2$ O $_3$ ) and then react with the SiO $_2$  on the substrate to form mullite. Mullite is desirable because its thermal expansion lies between that of zirconia and Si $_3$ N $_4$  or SiC.

Si was selected because its reactivity with other metals could improve the chemical bonding of the zirconia coating (which has excess zirconium on the bond interface).

Mo was deposited anticipating the formation of  $MoSi_2$  which forms a very stable interface with carbides. Also,  $MoSi_2$  has slow oxidation kinetics and, therefore may produce a stable interface with the  $ZrO_2$  coating.

Ti was selected for several reasons. Titanium dioxide (TiO<sub>2</sub>) forms a very stable interface with EB-PVD ZrO<sub>2</sub>, titanium carbide (TiC) forms a very stable interface with carbides, and titanium nitride (TiN) forms a stable interface with nitrides. Also, Ti-silicides have a melting point higher than both constituents and have excellent oxidation resistance.

After pretreatment, the substrates were EB-PVD coated with 75 to 125 microns of zirconia stabilized with 20 percent yttria\*. The coatings were applied at 980C. The initial few microns were deposited under oxygen deficient conditions (excess Zr) resulting in the thin, dense, equiaxed ZrO<sub>2</sub> coating desired for bonding. Oxygen was then introduced into the coating chamber resulting in a columnar microstructure with intercolumnar porosity, a structure which had an improved strain tolerance and compliance. The EB-PVD Y<sub>2</sub>O<sub>3</sub> stabilized ZrO<sub>2</sub> microstructure is illustrated in Figure 2.

On visual inspection of the as-coated specimens, the EB-PVD coatings appeared adherent. A typical as-coated specimen is illustrated in Figure 3. The coatings varied in color, ranging from white to blue

<sup>\*</sup>Coatings were applied at Temescal, Berkeley, California.

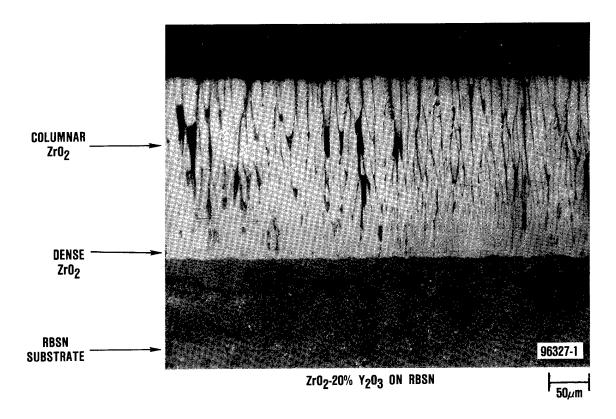


Figure 2. The EB-PVD zirconia coating grades from dense and microgranular at the interface to porous and columnar as it increases in thickness.

in Figure 3. The coatings varied in color, ranging from white to blue or gray. The color difference is typically associated with the stoichiometry of the ZrO<sub>2</sub> coating; white being stoichiometric and blue or gray being non-stoichiometric (oxygen deficient). In most cases the coatings were smooth and free of surface defects. Some specimens did exhibit small pits or bumps on the coatings which were attributed to "spitting" from the molten zirconia pool during coating. In some cases, the gray coatings were cracked or crazed. This effect could not be connected with any substrate, pretreatment or coating run.

Prior to evaluating adherence, some of the specimens received static air heat treatments at 1200C for 24 hours to evaluate the effect of oxidation exposures on the various coating/substrate systems. Visual inspection showed all coatings to be white regardless of their color prior to oxidation, indicating that all coatings were stoichiometric after oxidation exposure. For most systems, other than color change, little change in the appearance of the coatings was noted. The only significant change noted was that the ZrO<sub>2</sub> coatings on the Mo interlayer specimens cracked and spalled during heat treatment (Figure 4). Initially, it was felt that Mo was forming volatile oxides. Therefore, spare Mo interlayer specimens were heat treated in vacuum to eliminate oxidation effects. However, both the Mo and the

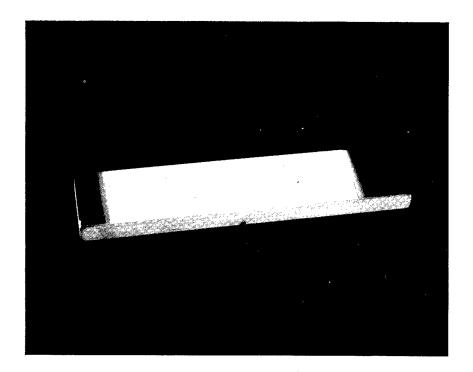


Figure 3. The EB-PVD zirconia appeared adherent to the substrates in the as-coated condition.

 ${\tt ZrO_2}$  spalled off together in sheets. Based on these results, the cause of spalling appears to be due to a thermal expansion mismatch or due to Mo reacting with the substrate and/or the coating. For all other coating/substrate systems, the  ${\tt ZrO_2}$  coating appeared clean and adherent after oxidation.

### 3.1.2 Adherence Testing

3.1.2.1 Adherence Assessment Procedures. The coating adherence was evaluated during Task 1 using three techniques: Scratch testing, flexure testing, and indentation testing.

Scratch testing was used primarily as a screening test. The test consists of sliding a moderately sharp implement across the coated surface. Visual observations were used to grade adherence. Coatings with poor adherence easily debond and flake off in sections. Moderately-adherent coatings require higher pressure to induce debonding, while adherent coatings show little or no debonding and only abrade.

Flexure testing was performed on as-coated specimens to examine the effects of mechanical shock on the various coating/substrate systems. Observation of the coating integrity in the contact area under the flexure load pins also was used to assess coating adherence. Flexure tests also provided preliminary evaluations of potential substrate degradation due to the coating process. Flexure tests were conducted such that the coated side of the specimen was in tension. An Instron Universal Testing Machine was used to load the specimen in four-point bending. The inner and outer spans of the test fixture

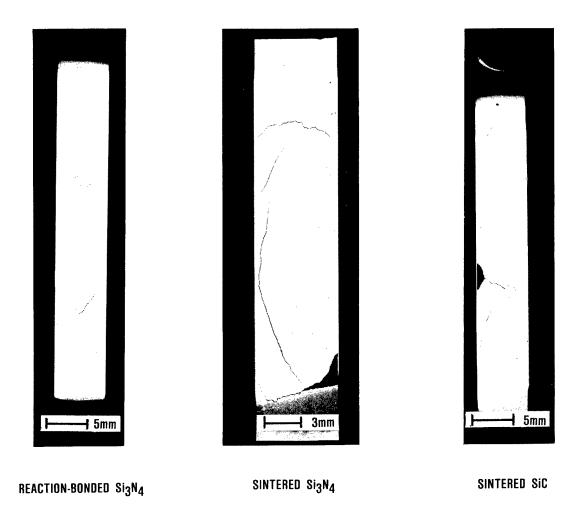


Figure 4. Molybdenum interlayer specimens exhibited severe coating cracking during the ZrO<sub>2</sub> coatings on oxidation exposure.

were 12.7 and 25.4 mm, respectively. The Instron crosshead speed was 0.5 mm per minute. After testing, all fracture surfaces were examined up to 40% using an optical binocular microscope. Visual observations were used to assess coating adherence. A coating with "good" to "excellent" adherence will remain attached to all fractured pieces of the substrate. A weakly-bonded coating will usually spall adjacent to the fracture and under the flexure load pins. The strength of the specimen was considered in the adherence assessment since stronger specimens are subjected to higher loads and a more severe mechanical shock during testing.

Indentation testing\* was also used to assess coating adherence. Testing consisted of applying a Vicker's indent normal to the coating. The Vicker's indenter is a sharp four-sided pyramid which produces

<sup>\*</sup>Indentation tests were conducted at The University of California, Berkeley under the direction of Dr. A.G. Evans.

deformation in the coating and a wedging or buckling action at the coating/substrate interface. Four indentations were applied to each specimen at loads of 5, 10, 15, and 20 kg. The average radius of debond was used to assess coating adherence. The coating debond area was clearly visible and could be measured. Indentation is preferred to the methods described above since it is a more controllable test and gives quantitative information.

3.1.2.2 <u>Adherence Test Results</u>. In general, the adherence results from scratch, flexure, and indentation tests were in agreement. The trends observed are summarized below:

### As-Coated Adherence

- o Relatively good adherence was observed for most coating/substrate systems.
- o RBSN substrates exhibited better coating adherence than SSN and SSC.
- o As-machined substrates yielded better adherence than lapped surfaces for SSN and SSC; no difference was observed for RBSN.
- Oxidation and etching pretreatments had no significant effect on coating adherence.
- o All sputter-deposited interlayers tended to decrease coating adherence.

### Adherence After Oxidation

- The coating adherence for all coating/substrate systems was reduced significantly relative to the as-coated adherence.
- o RBSN substrates exhibited better coating adherence than SSN and SSC.
- o As-machined substrates yielded better adherence than lapped surfaces.
- o Pre-oxidation and etching pretreatments had no significant effect on coating adherence.
- o Sputter-deposited interlayers yielded poor coating adherence.

The results obtained using each of the adherence tests are discussed individually in the following paragraphs.

Scratch testing revealed a range of adherences for the various coating/substrate systems. As-coated RBSN, with pre-oxidation or etching pretreatments, exhibited the best adherence. Scratching with moderate pressure did not cause debonding and resulted only in minor abrasion of the coating surface. Heavy pressure was required to cause debonding. The degree of adherence appeared to be less for as-coated SSN and SSC with the same pretreatments, but could still be classified as good. All three substrate materials exhibited less adherence when a sputtered interlayer was applied. Coatings on sputter-coated substrates debonded with moderate pressure of the test implement. Scraping could then cause the locally-debonded coating to spall off. Substrates with machined finishes tended to yield better coating

adherence than lapped substrates. This trend appeared to be independent of the substrate or pretreatment.

After oxidation exposure at 1200C for 24 hours, the coating adherence was reduced in all cases. Although the adherence loss was significant for all coating/substrate systems, the general trends observed for as-coated systems still existed. RBSN exhibited better coating adherence than SSN and SSC substrates; machined surfaces yielded better adherence than lapped; and interlayer specimens yielded the poorest adherence. As reported in Section 3.1.1, the Mo interlayer specimens cracked and spalled on heat treatment. Specimens with pre-oxidation heat treatments generally appeared to have slightly better retained coating adherence. Heavily pre-oxidized RBSN exhibited the best overall adherence after oxidation exposure, although the level of adherence was much less than observed for the as-coated specimens.

Since oxidation exposure had a detrimental effect on coating adherence, studies involving heat treatments of 1200C for 100 hours and 1400C for 24 hours were conducted to obtain additional information. The adherence characteristics observed after 100 hours of exposure at 1200C were similar to those observed after 24 hours at the same temperature. RBSN still exhibited the best adherence; machined substrates gave better adherence than lapped; and interlayer specimens exhibited the poorest adherence. The results suggested that the interface may stabilize within the first 24 hours of heat treatment. After oxidation exposure at 1400C for 24 hours, a slight reduction in coating adherence relative to the 1200C exposures was observed. ever, some differences in the adherence trends were noted. SSC systems now appeared to have better adherence than RBSN and SSN. the adherence loss for Si interlayer specimens after the 1400C exposure was not as significant as seen after the 1200C exposures (though the adherence loss relative to the as-coated condition was still significant).

Flexure testing revealed the same general adherence trends as identified using scratch testing. RBSN substrates exhibited the best coating adherence regardless of surface preparation or pretreatment. In most cases, the as-coated ZrO<sub>2</sub> remained adhered to the RBSN substrates up to the fracture surface (Figure 5) with no debonding or spalling. In addition, the coatings adhered to SSN and SSC, however some debonding and chipping at the fracture surfaces did occur more frequently with the SSN material (Figure 6).

Machined surfaces exhibited better coating adherence than lapped surfaces. Coatings on lapped substrates frequently debonded under the flexure load pins.

The metallic interlayer pretreatments reduced coating adherence relative to the baseline (specimens with no pretreatment). The worst adherence was observed for metallic interlayers on specimens with lapped substrates. These specimens frequently exhibited spalling (Figure 7).

The strength results, summarized in Table 3, suggest that the pretreatment and coating procedures do not significantly affect the

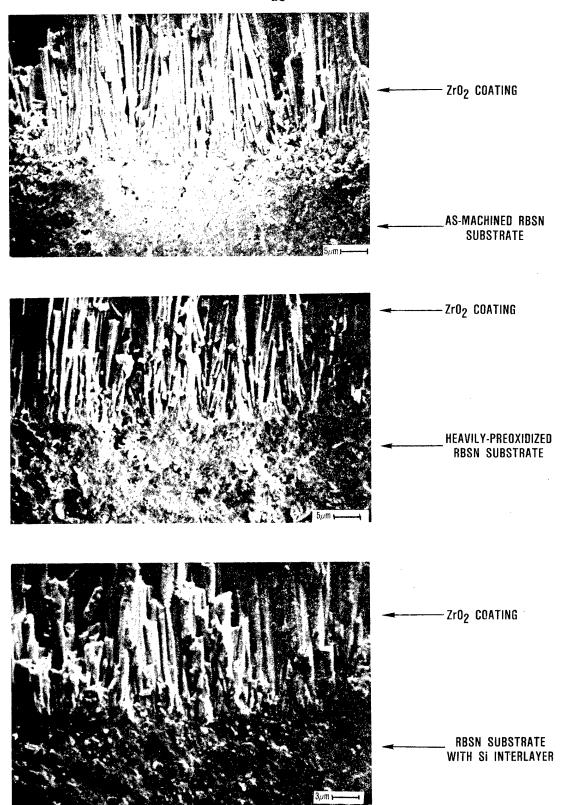
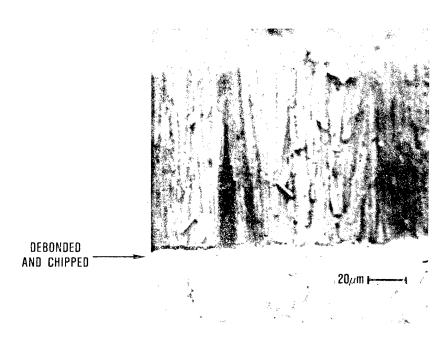
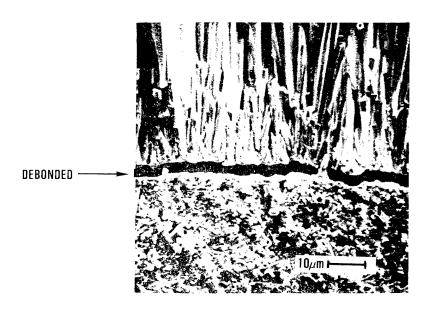


Figure 5. As-coated zirconia remained adherent to RBSN substrates with various pretreatments through flexure testing.



 ${\sf Zr0_2}$  COATING ON AS-MACHINED SINTERED  ${\sf Si_3N_4}$ 

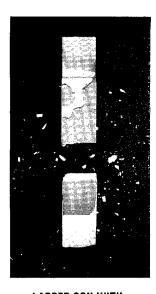


ZrO2 COATING ON SINTERED Si3N4 WITH A SI INTERLAY

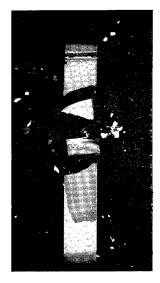
Figure 6. In some instances, flexure testing resulted in debonding and/or chipping of zirconia coatings on sintered silicon nitride substrates.







LAPPED SSN WITH TI INTERLAYER



LAPPED RBSN WITH Mo INTERLAYER

Figure 7. The worst zirconia adherence was observed for lapped substrates with metallic interlayers.

Table 3. Comparison of the strength of coated test bars with GTEC's baseline data on substrate strength.

Substrate	Average baseline strength MPa (ksi)	Average coated strength MPa (ksi)
RBSN	364.0 ±60.0	306.1 ±41.4
	(52.8 ±8.7)	(44.4 ±6.0)
SSN	582.6 ±76.5	533.0 ±90.3
	(84.5 ±11.1)	(77.3 ±13.1)
SSC	382.0 ±51.4	371.6 ±66.2
	(55.4 ±7.3)	(53.9 ±9.6)

substrate strength. The slightly reduced strengths of RBSN and SSN were probably coincidental since the substrates for Task 1 were selected from spare stock.

The <u>indentation test</u> results are summarized in Figure 8, where the average radius of the debonded area is plotted as a function of the indentation load. The range of debonding observed is illustrated in Figure 9. The results are in agreement with those obtained by scratch testing and flexure testing. The best coating adherence was observed for RBSN (both machined and lapped) with pre-oxidation or

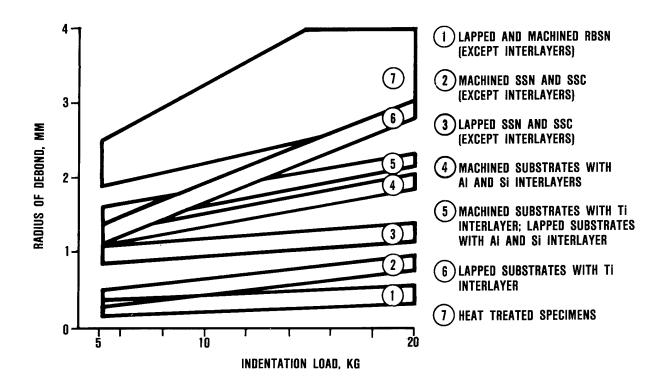


Figure 8. Indentation results are summarized plotting the radius of debond versus indentation load. The highest curve (#7) represent poorest observed, the lowest curve (#1) represents the best adherence.

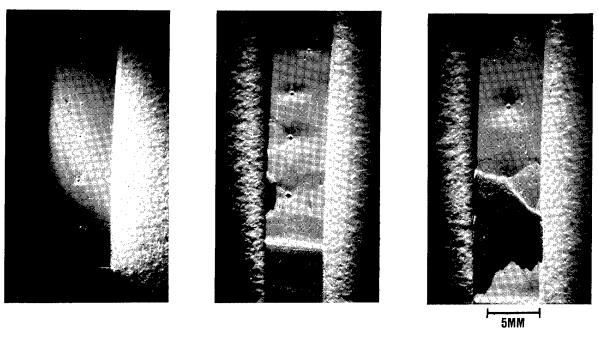


Figure 9. Indentation testing provided the range of debonding observed above.

etching pretreatments. SSN and SSC as-machined substrates yielded slightly less adherence. Lapping the SSN and SSC substrates resulted in lesser adherence than for as-machined SSN and SSC. The sputtered metallic interlayers degraded as-coated adherence for all three sub-Titanium yielded poorer coating adherence than strate materials. The  ${\rm ZrO}_2$  coatings adhered better to interlayers aluminum or silicon. sputtered on as-machined substrates. After oxidation exposure at 1200C for 24 hours, all coated specimens, regardless of substrate or pretreatment, had poor coating adherence. In many cases, the heattreated specimens exhibited debonding that extended to the edges of the specimen (Figure 10) so that the debond area could not be quantified accurately to be included in Figure 8. Therefore, the degree of adherence loss on oxidation exposure is greater than illustrated in Figure 8.

Indentation testing was effective in assessing the adherence of ZrO2 coatings before heat treatment. However, this technique was not

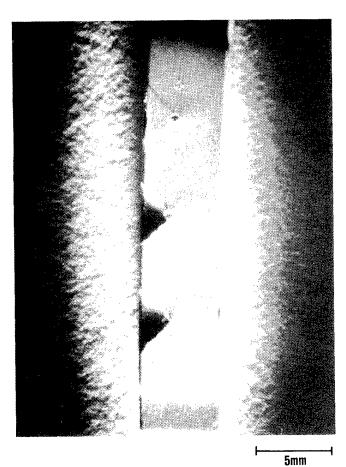


Figure 10. Many-heat treated specimens exhibited debonding that extended to the edges of the specimen and therefore did not provide comparative information.

as useful for quantitatively characterizing the weakly adherent coatings after heat treatment since extensive debonding and chipping occurred during testing. Evidently, the localized stress conditions imposed by the Vicker's indenter are too severe for evaluating the weaker coatings. In addition, they appear to be much more severe than likely to occur in a heat engine. A test using less severe stress conditions would compliment the indentation results in evaluating weaker coating/substrate systems. Also, conditions better simulating heat engine application conditions are desired for predicting the coating's performance in use.

# 3.1.3 Evaluation of Alternate Adherence Assessment Techniques

It was recognized in Task 1 that an alternate quantitative adherence assessment technique was needed for evaluating poorly adherent coating systems, as well as the adherent ones. Therefore, the development and evaluation of alternate methods of adherence assessment which better simulate application conditions was initiated. Tests considered were those which enabled the load distributing capabilities of the low-modulus EB-PVD coating to control the stresses present at the coating/substrate interfaces. Two techniques were evaluated: Blunt indentation testing and line contact\* testing.

3.1.3.1 <u>Blunt Indentation Testing</u>. Blunt indentation testing involves procedures similar to those of Vicker's indentation testing but involves the use of a 3.175-mm-diameter sapphire ball indenter. The sharp Vicker's indenter severely deforms the coating and penetrates to the coating/substrate interface during testing. The ball indenter produces a less severe stress concentration, is closer to conditions likely to occur in a heat engine, and is less likely to cause the large degree of debonding observed with Vicker's indentation.

As expected, results from ball indentation suggested a less severe stress concentration. However, other than deformation of the coating from the ball indenter, no visual damage resulted, leaving no means for assessing the coating adherence (Figure 11). Modifying this technique to induce damage would involve the use of a smaller diameter ball. Because this modification was a step back toward Vicker's indentation, efforts in developing new indentation procedures were not continued.

3.1.3.2 <u>Line Contact Testing</u>. Line contact testing involved subjecting the coating to biaxial stress conditions by displacing a line contact load across the coating surface. Figure 12 illustrates the test rig used for contact testing. With this apparatus, a normal force is applied to the specimen through a dead-weight load system

<sup>\*</sup>Line contact describes the contact geometry between a cylinder and a flat plate.

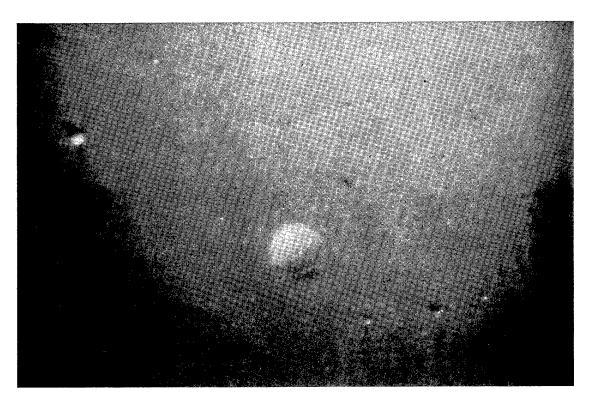


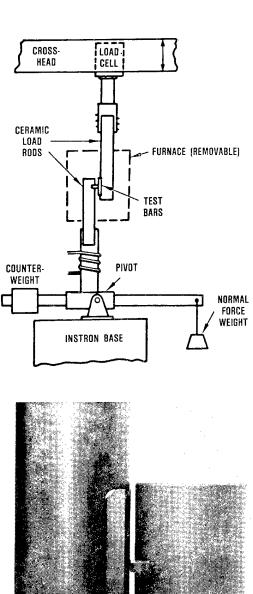
Figure 11. A 20 kg ball indentation provides no information for assessing coating adherence. No damage, other than deformation of the coating, occurred.

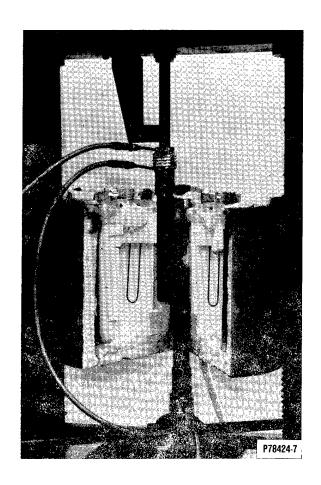
while a tangential force is applied through displacement of the Instron crosshead. The tangential force is monitored by a load cell located in the crosshead. This mode of loading was identified in prior studies to be a critical condition to evaluate. Contact testing was originally planned for use in Task 3 of the program for determining friction characteristics and the resistance to contact damage. However, exploratory tests indicated that line contact testing is also useful for coating adherence characterization.

Contact tests were conducted in the static mode (i.e., normal and tangential loads are applied but no sliding was initiated) and in the sliding mode (i.e., normal and tangential loads are applied where the tangential load was sufficient to cause sliding).

Static tests were conducted using up to 27.2 kg normal load, with tangential loads 20 percent of the normal (preliminary tests suggested 25 to 30 percent of the normal load is required to initiate sliding). No significant damage to the coatings occurred under these test conditions, as illustrated in Figure 13. In no case did the coating spall or chip regardless of substrate material or pretreatment. A few specimens exhibited debonding at the highest loads, but the debonding was contained to the thin line of contact.

Emphasis shifted to sliding contact tests since static tests did not generate significant damage to the coatings. For sliding contact





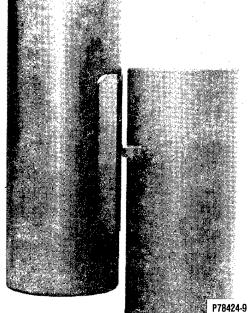


Figure 12. The GTEC Contact Test Rig applies a normal load through a dead weight loading system while a tangential force is applied through displacement of the Instron crosshead.

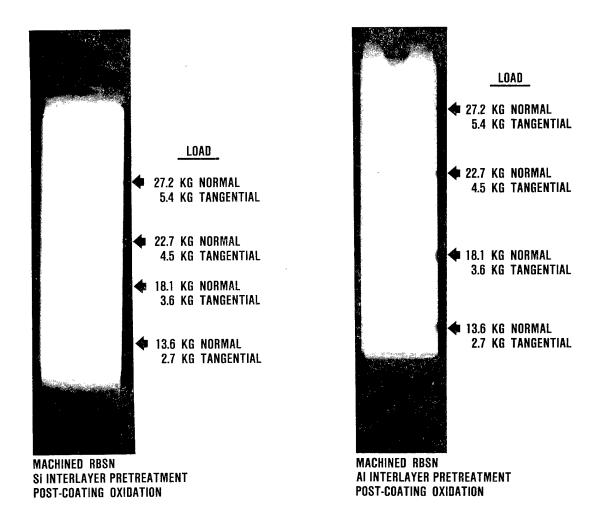
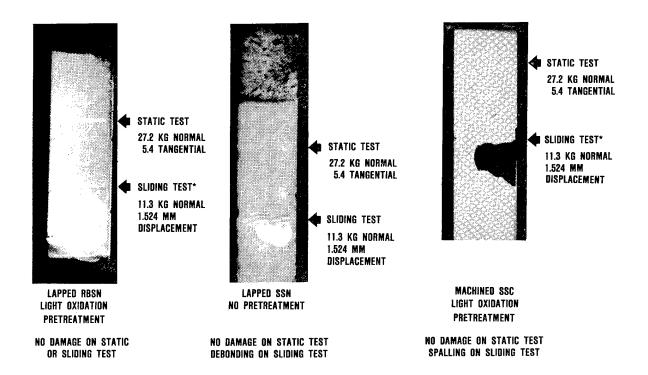


Figure 13. No significant damage to the coatings occurred during static line contact testing. The specimens shown above have relatively poor coating adherence based on results form scratch, flexure, and indentation tests.

tests, a normal contact load was displaced 1.52 mm. An 11.3-kg normal load was used in this study. Results suggested this test can be used for coating adherence evaluation. Figure 14 compares the effects of static and sliding contact test conditions on coated specimens. In most cases, sliding tests using a 11.3 kg normal load could induce damage, while static tests using a 27.2 kg normal load could not.

From the contact test results, it was concluded that sliding line contact should be used to complement the indentation test in future adherence assessment tasks. The preliminary contact tests showed that sliding line contact is an attractive technique for adherence assessment since it is amenable to the poorly adherent coatings. Also, the sliding contact test may better assess which systems would perform best in a contact stress application since the test simulates contact conditions at ceramic component interfaces.



### \*DIRECTION OF CONTACT SLIDING MOTION

Figure 14. In most cases, sliding tests using 11.3 kg normal load could induce damage where static tests using 27.2 kg normal load did not.

### 3.1.4 Interface Analysis

Interface analyses\* were performed to determine the nature of the coating adherence degradation observed on oxidized specimens. Since zirconia is permeable to oxygen, oxidation of the substrate was anticipated. Polished specimens were prepared so that coating/substrate interfaces could be examined using a high-resolution scanning electron microscope (SEM). Energy-dispersive X-ray (EDX) analysis was used to provide chemical information.

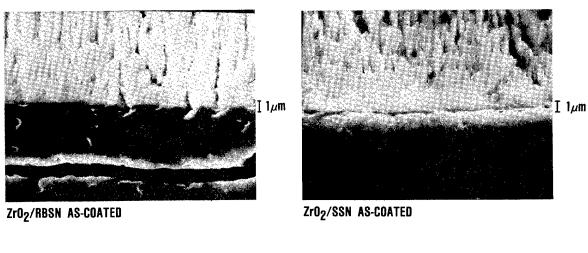
Four specimens were prepared for evaluation: ZrO2/RBSN and ZrO2/SSN both before and after oxidation exposure. The SEM results suggested that a silicate layer at the interface was responsible for the adherence loss. For as-coated ZrO2/RBSN and ZrO2/SSN, a thin layer was detected at the coating/substrate interface. It was suspected that this layer was a zirconium silicate since the major constituents detected by EDX were Si and Zr. The results suggest that ZrO2 coatings adhere to the silicate since most as-coated specimens without sputtered interlayers exhibited relatively good adherence. This silicate layer probably formed during coating since oxidizing

<sup>\*</sup>Interface analyses were conducted at The University of California, Berkeley, under the direction of Dr. A.G. Evans.

conditions existed (i.e., oxygen was bled into the coating chamber during deposition). The micrographs shown in Figure 15 indicate significant growth of the silicate layer during subsequent oxidation. For ZrO<sub>2</sub>/RBSN, the silicate layer (not visible for the as-coated specimen) grew to approximately 1 micron in thickness during heat treatment. Some porosity appears to be present within the silicate but this may have resulted from pullout during polishing. The hypothesis for the adherence degradation observed is that the silicate layer after growth is cohesively weak. Similar effects were observed for the ZrO<sub>2</sub>/SSN system; the silicate layer grew between 2 and 3 microns in thickness. The thicker silicate layer may be a result of the sintering additives present in SSN.

# 3.1.5 Task 1 Summary and Conclusions

The EB-PVD process produced controlled coatings of  $\rm ZrO_2$  20-percent  $\rm Y_2O_3$  with good to excellent adherence on machined surfaces of



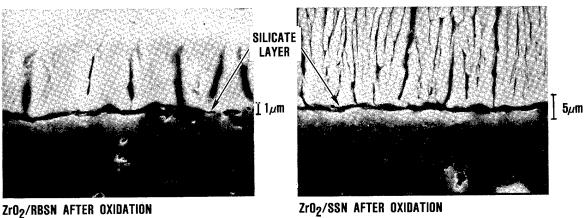


Figure 15. Oxide growth at the coating substrate interface takes place during oxidation exposure.

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RBSN, SSN, and SSC. The coating procedures appeared not to have any detrimental effects on substrate strength. SEM and EDX results suggested that the bond interface between the  $\rm ZrO_2$  and as-machined substrate is a thin film of zirconium silicate.

Mechanical adherence contributed significantly to the overall adherence of EB-PVD ZrO<sub>2</sub>. RBSN, which has the roughest surface of the substrate materials due to its 15-percent residual porosity, consistently exhibited better adherence than the dense SSN and SSC substrates. Also, lapping the 320-grit ground surfaces prior to coating resulted in poorer adherence in most cases.

Of the various oxidation, etching, and sputter coating pretreatments, none imparted any additional adherence to the as-coated specimens. The oxidation and the etching pretreatments did not have any significant effect on the as-coated adherence relative to baseline substrates. The sputtered interlayers (aluminum, silicon, molybdenum, and titanium) decreased the level of coating adherence relative to the baseline.

Static air oxidation exposures of coated specimens significantly degraded the coating adherence for all coating/substrate systems. The loss of adherence was attributed to the growth of the silicate layer at the coating substrate interface. Since  ${\rm ZrO_2}$  is known to be permeable to oxygen, it was anticipated that oxidation of the substrates would occur in cases where the  ${\rm ZrO_2}$  was deposited on bare  ${\rm Si}_3{\rm N}_4$  or  ${\rm SiC}$  substrates. Since the oxidation and sputtered interlayer pretreatments were unsuccessful in preventing adherence loss, it was concluded that approaches aimed at reducing or eliminating adherence loss due to oxidation should be emphasized in subsequent tasks.

Vicker's indentation was an effective technique for characterizing strongly adherent coatings. However, the interfacial stress conditions imposed by the Vicker's indenter were too severe for the weakly bonded coatings observed after oxidation. For oxidized specimens, extensive debonding and chipping occurred on testing. Therefore, alternate adherence assessment methods utilizing less severe stress conditions were screened during Task 1 for potential use in later adherence evaluations. Subsequently, test procedures for line contact testing were developed which are applicable to both poorly and well-bonded coatings. Using both contact and indentation testing was expected to be more appropriate for coating adherence assessment than using only indentation testing.

### 3.2 TASK 2 - ADVANCED PRETREATMENT AND COATING STUDIES

During Task 2, developmental coating and adherence assessments continued. Adherence evaluation was modified to include contact testing, and emphasis was placed on coating adherence after post-coating oxidation heat treatment. Based on Task 1 observations, directions for adherence improvements were defined for Task 2. Primary approaches were the use of an oxygen diffusion barrier at the coating/substrate interface to inhibit oxide growth and the use of surface

roughening techniques to improve the coating's mechanical bond. Based on Task 2 adherence assessment results, the best coating approaches evaluated were selected for assessment of contact damage resistance in Tasks 3 and 4.

### 3.2.1 Selection of Coating Approaches

The results of Task 1 provided two major conclusions concerning the direction for adherence improvements:

- o The growth of a silicate layer at the 2rO<sub>2</sub>/substrate interface during oxidation exposure significantly degrades coating adherence.
- o Mechanical bonding to rougher surfaces contributes to coating adherence.

Therefore, Task 2 efforts were directed toward the development of oxygen diffusion barriers to inhibit interface oxidation, and toward surface preparations to improve the mechanical adherence of the coatings. In addition to these efforts, the use of high-purity interlayers, diffusion/gradation zones, higher coating temperatures, and mullite coatings were evaluated. These approaches are discussed in the following paragraphs.

3.2.1.1 Oxygen Diffusion Barriers. The objective of these efforts was to prevent substrate oxidation by depositing an oxygen diffusion barrier between the substrate and the coating. A secondary objective was to select an interlayer with good adherence to both the substrate and the coating. The candidate interlayers selected were CVD alumina, CVD aluminum nitride (AlN), EB-PVD alumina, EB-PVD mullite, and sol-gel alumina.

Three of the interlayers were alumina, each deposited by a different technique. Alumina is an attractive candidate due to its low oxygen permeability ( $10^{-14}$  g/cm/sec at 1200C). Also, the results from a study involving EB-PVD  $\rm ZrO_2$  coatings on sapphire showed that EB-PVD  $\rm ZrO_2$  bonded well to  $\rm Al_2O_3$ , and that adherence was retained through oxidation and cyclic thermal exposures.

CVD aluminum nitride was selected since it may form a stable bond with the substrates (particularly RBSN and SSN) while providing oxidation resistance (the surface of the AlN should oxidize to form alumina and provide the oxygen diffusion characteristics described above).

EB-PVD mullite is an attractive interlayer since it combines low oxygen diffusion with a thermal expansion coefficient between that of  $ZrO_2$  and  $Si_3N_4$  or SiC.

3.2.1.2 <u>Surface Roughening</u>. Various techniques for roughening the substrate surface were explored since substrate surface topography influenced coating adherence during Task 1. Candidate techniques included 150-grit diamond grinding (compared with 320-grit), hydrofluoric acid (HF) etching, and laser-texturing. Diamond grinding the substrate surface with a 150-grit increased the depth of the machining grooves; HF etching removed surface silica from the substrate to

potentially exaggerate surface features; and laser-texturing produced a matrix of cavities on the substrate surface.

- 3.2.1.3 <u>High-Purity Interlayer</u>. The use of a high-purity CVD  $Si_3N_4$  or SiC interlayer between the substrate and the EB-PVD  $ZrO_2$  coating was also examined. This study was performed to assess the effect of sintering additives in the substrates on coating adherence and adherence degradation during oxidation.
- 3.2.1.4 <u>Diffusion/Gradation Zone</u>. Techniques for producing a gradation or interdiffusion zone between coating and substrate were explored in order to chemically root the zirconia coating to the substrate. Candidate techniques included ion implantation of aluminum or zirconium, and ion-mixing nitrogen with aluminum sputtered on substrate. Due to scheduling difficulties, only the latter was evaluated.
- 3.2.1.5 <u>Coating Temperature</u>. Studies on metal substrate coatings have determined that adherence can be improved by increasing coating temperature. Prior plasma spray studies suggest a similar result for ceramic substrates. Therefore the effect of coating at higher temperatures was evaluated.
- 3.2.1.6 <u>Mullite Coating</u>. Developmental runs were conducted to determine if a thick mullite coating could be deposited by EB-PVD and result in better adherence than  $ZrO_2$ . Mullite is a better oxygen barrier than zirconia and has a closer thermal expansion match to SiC and  $Si_3N_4$  substrate materials.

# 3.2.2 Specimen Preparation and Inspection

The three substrate materials, RBSN, SSN, and SSC, were procured in the form of billets. The billets were machined into test bars measuring 6.35 by 3.23 by 50.8 mm having longitudinally ground surfaces. The substrate was then pretreated (if required) prior to EB-PVD coating.

3.2.2.1 Oxygen Diffusion Barrier Studies. Under Oxygen Diffusion Barriers studies, interlayer coatings were procured from several sources:

CVD alumina coatings\* 2 to 2.5 $\mu$ m thick were applied at 920C. The microstructure consisted of nodular grains one to three microns in size (Figure 16). X-ray diffraction (XRD) of the as-coated specimens showed the coating to be a mixture of alpha and kappa phase Al<sub>2</sub>O<sub>3</sub>.

<sup>\*</sup>Applied at Kennametal's Philip M. McKenna Laboratory, Greensburg, Pennsylvania.

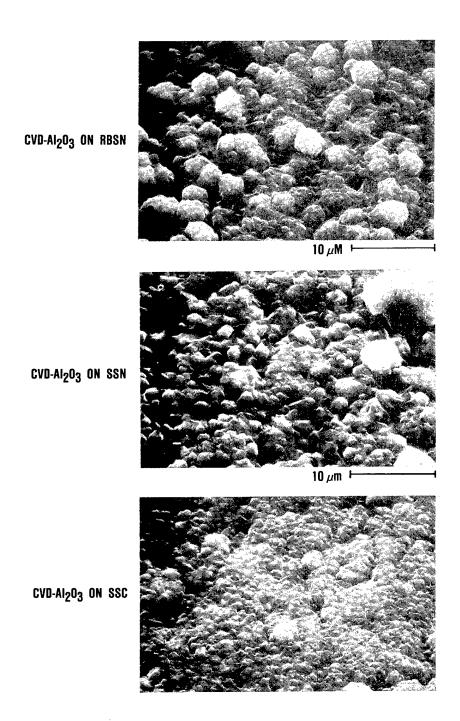


Figure 16. The microstructure of CVD alumina consists of nodular grains, one to three microns in size.

The specimens were heat treated in air at 1200C for 20 hours to convert the kappa-alumina to the alpha phase. XRD after heat treatment indicated complete conversion.

 $\underline{\text{CVD}}$  aluminum nitride coatings\* approximately 2 to 3 $\mu$ m thick were deposited using a fluidized bed technique.

<u>EB-PVD alumina</u> coatings\*\* approximately  $1\mu m$  in thickness were applied. The oxygen partial pressure in the coating chamber was varied to give two types of coating: stoichiometric  $Al_2O_3$  and non-stoichiometric  $Al_2O_3$ .

EB-PVD mullite coating\*\* runs were conducted. However, mullite coatings could not be successfully applied by EB-PVD. The silica in the mullite material source evaporated more easily than the alumina resulting in silica rich coatings. Efforts to obtain a mullite oxygen diffusion barrier were not pursued further.

Sol-gel alumina coatings\*\*\* were developed and applied. The specimens were dipped in alumina sol, air dried, then calcined in air at 1200C. The resulting coating was approximately 0.5µm thick. During development efforts, thicker coatings were pursued, but usually cracked and/or spalled on drying or heating.

3.2.2.2 <u>Surface Roughness Studies</u>. Under Surface Roughness studies, all substrate treatments were conducted at GTEC.

Using 150-grit diamond grinding, approximately 50 microns was ground off one surface of each test bar.

HF Etching studies were conducted to roughen the substrate surface through otching of residual surface oxides. As-machined and oxidized (1200C for 24 hours in air) substrates were etched 10 minutes using a 10 percent HF solution. The resulting surface topographies were compared to un-etched, as-machined substrates. The resulting surface topographies suggested HF etching should not significantly enhance mechanical adherence (Figure 17). Therefore, the HF etching studies were not pursued further.

Laser-texturing was conducted using a 1.2 kw CO<sub>2</sub> laser. The laser was operated in a pulsed mode at 20 to 30 percent of the rated power to produce a matrix of surface cavities. Preliminary experiments indicated that 0.1mm diameter cavities 0.3mm deep resulted in a 20 percent loss for the substrates. Subsequently, the laser power was decreased 35 percent to reduce the cavity depth. The cavity depth obtained for SCC was reduced significantly (cavities were about 0.1mm deep). However, RBSN and SSN exhibited little change in cavity depth. Specimens from this laser run were subsequently coated with ZrO<sub>2</sub> for adherence evaluations.

<sup>\*</sup>Applied at G.A. Technologies, San Diego, California.

<sup>\*\*</sup>Applied at Temescal, Berkeley, California.

<sup>\*\*\*</sup>Applied at Allied-Signal Engineered Materials Research Center, Des Plaines, Illinois.

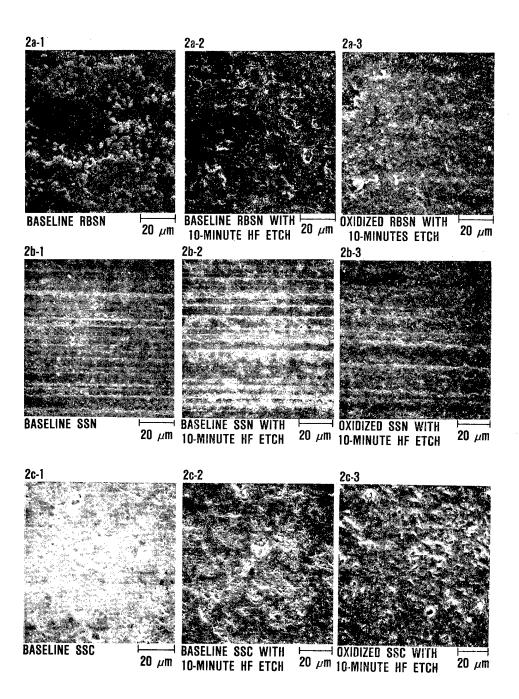


Figure 17. The surface topographies resulting from HF etching should not significantly enhance the mechanical adherence of the zirconia coating relative to baseline.

- 3.2.2.3 <u>High Purity Interlayer Studies</u>. In-house test bars of SSC have CVD SiC coatings\* were used in preliminary evaluations. Efforts were not extended past preliminary investigations because of initial discouraging results (see Section 3.2.3.2).
- 3.2.2.4 <u>Diffusion Gradation Zone Studies</u>. A thin layer of aluminum was sputtered on each substrate then ion mixed with nitrogen ions\*\*. Ion mixing was performed to drive aluminum atoms into the substrate for better bonding. Based on Task 1 results, aluminum was not expected to provide adequate bonding. Therefore, these specimens were heat treated 8 hours at 1200C in air in an effort to form an Al<sub>2</sub>O<sub>3</sub> layer before subsequent EB-PVD coating.
- 3.2.2.5 <u>Coating Temperature and Mullite Coating Studies</u>. No pretreatment was required.
- All specimens, except those under 3.2.2.6 EB-PVD Coating. Coating Temperature and Mullite Coating Studies, were coated with EB-PVD zirconia using the procedures described in Section 3.1.1. Coating Temperature Studies, the coating chamber was maintained at 1040C (the maximum temperature attainable by the equipment used) rather than 980C. Under Mullite Coating Studies, EB-PVD coating procedures for depositing zirconia were used. However, EB-PVD mullite could not be applied successfully to the substrates. The material source contained 80 weight percent alumina and 20 weight percent Dense, non-columnar, silica rich films resulted from coating runs conducted at temperatures ranging from 700C to 1010C (Figure 18). The SiO2 rich films probably resulted from the tendency of SiO2 to evaporate more easily than the Al<sub>2</sub>O<sub>3</sub>. Efforts involving EB-PVD mullite coatings, as well as EB-PVD mullite oxygen diffusion barriers, were terminated.

The color of the as-deposited ZrO<sub>2</sub> coatings were either white or blue (no gray coatings, which were seen occasionally in Task 1, resulted). All coatings appeared as clean and adherent as those deposited during Task 1. Some of the coated specimens from each coating/substrate variation were heat treated 100 hours at 1200C to stabilize the interface prior to adherence investigations. Additional heat treatments at 1200C and 1400C for 24 hours were conducted under Coating Temperature Studies to compare with oxidation results from Task 1.

#### 3.2.3 Adherence Testing

3.2.3.1 Adherence Assessment Procedures. Three techniques were utilized for coating adherence evaluations in Task 2: flexure testing, indentation testing, and line contact testing.

<sup>\*</sup>CVD SiC coatings were applied at San Fernando Laboratories, Pacoima, California.

<sup>\*</sup>Aluminum sputtering and ion mixing was performed at Oak Ridge National Laboratories (ORNL), Oak Ridge, Tennessee.



SILICA RICH COATING





**RBSN SUBSTRATE** 

AI MAP

SI MAP

Figure 18. EB-PVD coating runs using 80 weight percent Al<sub>2</sub>0<sub>3</sub> and 20 weight percent SiO<sub>2</sub> as a material source resulted in thin, dense, non-columnar, silica rich films.

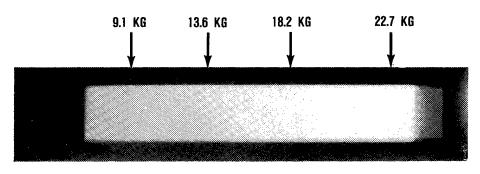
Flexure testing was conducted on as-coated specimens using the procedures described in Section 3.1.2.1 for Task 1 adherence assessment except that test fixture inner and outer spans of 19.05 and 38.1 mm, rather than 12.7 and 25.4 mm, were used since longer test bars were available.

Indentation testing was conducted on both as-coated and heat-treated specimens using the Task 1 procedures described in Section 3.1.2.1, except that loads of 5, 10, 20, and 30 kg were used instead of 5, 10, 15, and 20 kg. Extending the test range to higher loads was expected to provide a better assessment of more adherent coatings. It was anticipated that very adherent coatings would exhibit little measurable debonding when tested at the lower loads.

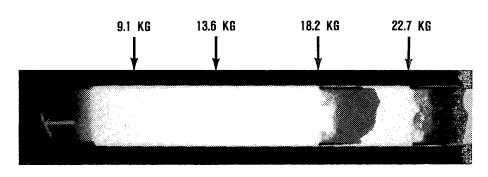
Line contact testing was used to evaluate the adherence of coated specimens having post-coating heat treatments. The contact rig was used for this testing and its operation is described in Section 3.1.3.1. Four tests were conducted per specimen with loads of 9.1, 13.6, 18.1, and 22.7 kg. The crosshead displacement was 1.52 mm. The threshold normal load required for debonding, determined through visual observations, was used to assess coating adherence. Typical specimens from line contact testing are illustrated in Figure 19.

#### 3.2.3.2 Adherence Test Results.

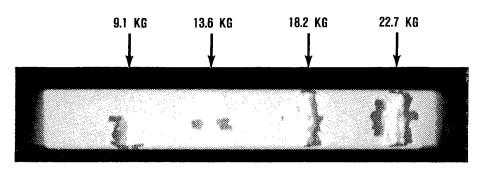
Oxygen Diffusion Barriers. CVD Al<sub>2</sub>O<sub>3</sub> and CVD AlN interlayers yielded the best adherence for RBSN. All other interlayers resulted in lesser adherence. The best adherence characteristics for SSN and SSC involved sol-gel alumina interlayers, although the level of adherence was much less than that observed for the ZrO<sub>2</sub>/CVD Al<sub>2</sub>O<sub>3</sub>/RBSN and ZrO<sub>2</sub>/CVD Al<sub>N</sub>/RBSN systems. Other interlayers resulted in less adherence.



**EXCELLENT ADHERENCE** 



FAIR ADHERENCE



POOR ADHERENCE

Figure 19. After line contact testing, coating adherence is assessed using visual observations.

CVD alumina interlayer specimens were evaluated using indentation and line contact testing. The indentation results are shown in Figure The RBSN specimens exhibited excellent adherence before and after oxidation exposures of 100 hours at 1200C. However, the CVD Al2O3 interlayers did not improve the coating adherence for SSN and SSC (EB-PVD ZrO2 on as-machined substrates). SSN and SSC exhibited good adherence in the as-coated condition, but poor adherence after oxida-The line contact test results for CVD alumina interlayer specimens with post-coating oxidation exposures are shown in Excellent adherence was observed for RBSN; up to 22.7 kg normal load was used with no evidence of coating separation (12.7 kg Fair adherence was observed for SSN; the coating tangential load). spalled during the 18.1 kg test. Poor adherence was observed for SSC These results suggest CVD alumina does which debonded at 13.6 kg. inhibit degradation of the interface during oxidation exposure for RBSN. However, the CVD alumina does not appear to benefit SSN or SSC. The hypothesis for this difference is that thermal expansion mismatches between the substrate and the coating resulted in interfacial Due to the residual surface porosity of RBSN, islands of cracking. interlayer remained adherent due to rooting into the substrate pores.

CVD aluminum nitride interlayer systems were evaluated using flexure, indentation and line contact testing. Flexure testing was only conducted on as-coated specimens. Debonding and/or spalling occurred under the load pins for all tests suggesting poor coating

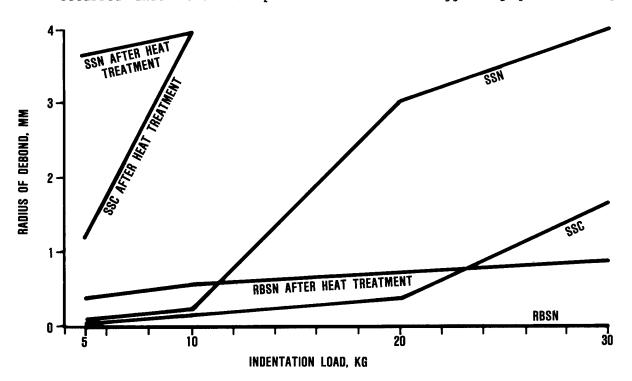


Figure 20. RBSN specimens with CVD Al<sub>2</sub>O<sub>3</sub> interlayers exhibited excellent adherence before and after oxidation heat treatment.

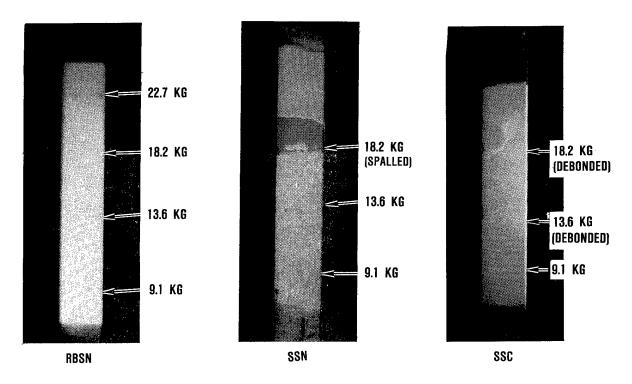


Figure 21. Contact testing results of oxidized CVD alumina interlayer specimens show excellent adherence for RBSN, fair adherence for SSN, and poor adherence for SSC.

adherence for all three substrate materials. EDX analysis of spalled surfaces suggested that the zirconia spalled from the AlN interlayer since a strong aluminum peak, and no silicon peak, was detected. Indentation tests were conducted on both as-coated and oxidized speci-The results are shown in Figure 22. The as-coated ZrO2 mens. adherence was poor for all three materials, which agreed with the flexure test results. However, the adherence improved significantly after oxidation exposure for RBSN and slightly for SSN and SSC. exhibited excellent adherence after oxidation. Fair adherence was observed for SSN and SSC. The better adherence observed for RBSN may have resulted from the CVD coating rooting into RBSN's residual porosity. Line contact testing was used to evaluate the adherence of oxidation exposed SSN and SSC specimens. Poor and fair adherence was exhibited by SSN (spalled at 9.1 kg normal load) and SSC (debonded at 13.6 kg, no spalling up to 22.7 kg) respectively. The ZrO2/CVD AlN/ RBSN specimens were oversize and could not be accommodated by the contact rig. Therefore, a specimen was evaluated by scratch testing using the procedures described in Section 3.1.2.1. The adherence observed was excellent as the coating could not be dislodged from the The coating would only abrade when scratched under heavy substrate. pressure. The increased adherence of the zirconia coating after oxidation exposure suggests chemical interaction between the ziconia coating and AlN interlayer. Optical, SEM and EDX analysis of the interface, before and after oxidation, was performed to determine the

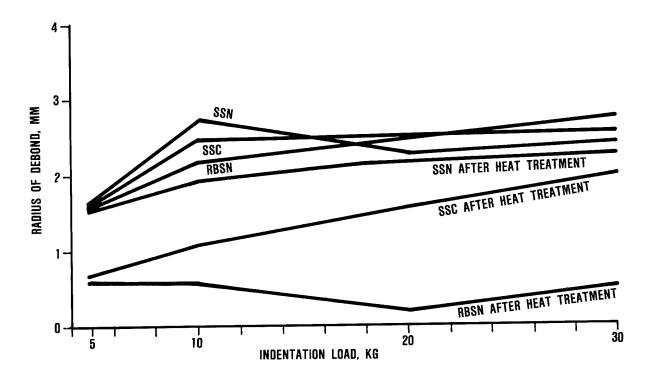
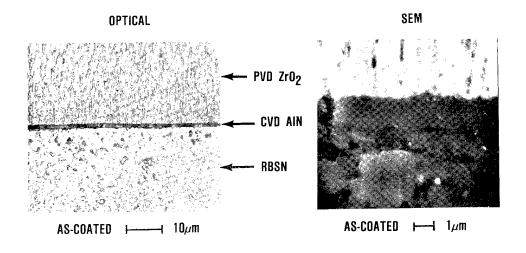


Figure 22. The adherence for ZrO<sub>2</sub>/CVD AlN/RBSN increased significantly after heat treatment.

nature of the adherence improvement. The optical and SEM results suggest a change in interfacial characteristics (Figure 23). However, EDX analysis did not suggest any diffusion took place during oxidation exposure. Evidently, the CVD AlN oxidized to form  ${\rm Al}_2{\rm O}_3$  which resulted in the excellent post-oxidation adherence characteristics.

Al<sub>2</sub>O<sub>3</sub> interlayer specimens (stoichiometric and nonstoichiometric) were tested for adherence using flexure, indentation, and line contact tests. For flexure tests, extensive chipping and/or debonding was observed for non-stoichiometric Al203 interlayer speci-EDX analysis suggests that the non-stoichiometric mens (Figure 24). Al2O3 interlayers were contaminated prior to EB-PVD ZrO2 coating since sodium (Na) and potassium (K) were detected on the still-adhered EB-PVD  ${\tt Al}_2{\tt O}_3$  interlayer (Figure 25). A lesser degree of chipping and debonding was observed for SSN and SSC with stoichiometric  ${\tt Al}_2{\tt O}_3$ RBSN with a stoichiometric Al<sub>2</sub>O<sub>3</sub> interlayer was not interlayers. damaged during flexure testing. EDX analysis of spalled areas on SSN specimens suggested that EB-PVD Al2O3 adhered poorly to SSN since primarily Si (from the substrate) was detected. The indentation results are shown in Figures 26 and 27. The ZrO2 coating adherence was poor for both Al<sub>2</sub>O<sub>3</sub> interlayers, but the stoichiometric Al<sub>2</sub>O<sub>3</sub> was slightly Contact testing conducted on oxidation-exposed specimens showed poor adherence for the non-stoichiometric interlayers but fair adherence for stoichiometric Al<sub>2</sub>O<sub>3</sub>. However, the tangential loads generated were lower than typical, resulting in a less severe stress condition.



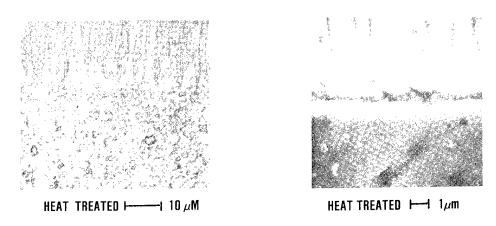


Figure 23. Optical and SEM analysis of the ZrO<sub>2</sub>/CVD AlN/RBSN interface before and after oxidation exposure suggests a change in interfacial characteristics.

Sol-gel Al<sub>2</sub>O<sub>3</sub> interlayer specimens were evaluated using flexure, indentation, and line contact testing, with varying results. Flexure test results suggested good adherence for RBSN and SSN (no spalling or debonding) and fair adherence for SSC (debonding under the flexure load pins). SEM micrographs of the fracture surfaces (Figure 28) show intimate contact at the interfaces for all three systems. In indentation tests with as-coated specimens, RBSN exhibited excellent adherence, while SSN and SSC exhibited fair adherence (Figure 29). After oxidation exposure, SSN and SSC exhibited no significant change in adherence characteristics, suggesting that the sol-gel Al<sub>2</sub>O<sub>3</sub> prevented degradation. The ZrO<sub>2</sub>/sol-gel Al<sub>2</sub>O<sub>3</sub>/RBSN coating adherence degraded significantly after oxidation. The retention in adherence observed for SSN and SSC was not expected since CVD Al<sub>2</sub>O<sub>3</sub> did not provide adherence retention for these two substrates. Speculation is

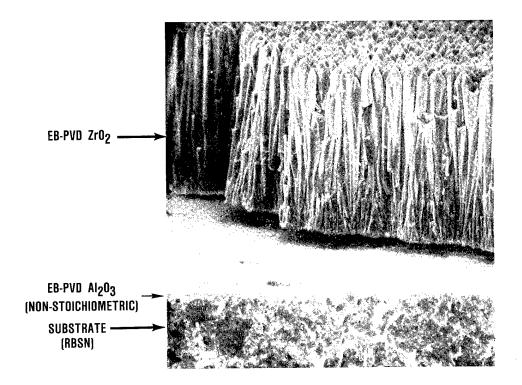


Figure 24. Zirconia coatings typically spalled from the non-stoichiometric EB-PVD alumina interlayers during flexure testing.

that the 50 percent porosity in the sol-gel interlayer (50A porosity) provides improved strain tolerance to limit interfacial cracking caused by thermal expansion mismatches. However, improved strain tolerance is provided at the cost of the interlayer's cohesive strength. Line contact testing on oxidized specimens suggested good adherence for all three substrates, which contradicts the above results. However, the tangential loads generated by friction were lower than typically observed and resulted in a less severe stress condition than used on other systems.

<u>Surface Roughening</u>. The results from surface roughening studies suggested that 150-grit diamond grinding has no significant effect on ZrO<sub>2</sub> coating adherence relative to 320-grit, and that laser texturing can significantly increase both the as-coated and oxidized adherence.

The 150-grit machined substrates were indentation and line contact tested for ZrO<sub>2</sub> coating adherence. Indentation results are shown in Figure 30. The as-coated adherence was good to excellent. After oxidation exposure, the adherence for RBSN, SSN, and SSC was poor. Contact tests were performed on oxidized specimens. RBSN and SSN exhibited fair adherence (spalled at 40 pounds) and SSC exhibited poor adherence (spalled at 30 pounds). The results suggest that 150-grit

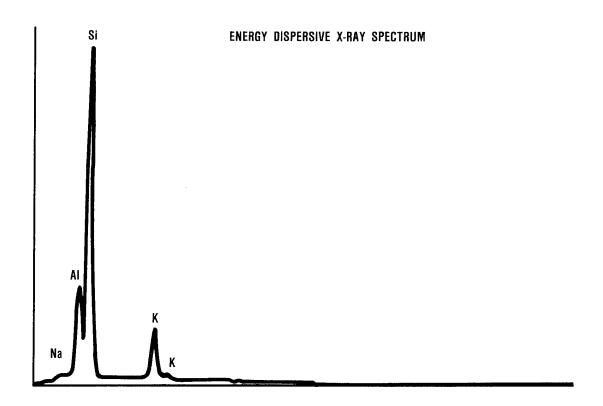


Figure 25. EDX analysis of spalled area suggests the non-stoichiometric alumina coatings were contaminated prior to EB-PVD zirconia application since sodium (Na) and potassium (K) were detected.

machining does not have beneficial effects on coating adherence relative to 320-grit machining.

Laser-textured specimens were evaluated for coating adherence using indentation and line contact testing. Indentation tests were conducted on as-coated and heat-treated specimens. The results show excellent 2rO2 adherence before and after oxidation (Figure 31). contact tests were performed on heat treated specimens. The results (Figure 32) did not completely agree with indentation test results. The results only agreed for SSC, which exhibited very good adherence (minor spalling at 22.7 kg). RBSN and SSN exhibited poor adherence (spalled at 13.6 and 9.1 kg, respectively). The differences observed in adherence were attributed to differences in the surface topography of the substrates and not to substrate material. RBSN and SSN had very deep surface cavities; analysis of spalled areas showed that these deep cavities were not successfully filled in with by the zirconia and that only the walls of the cavities were coated (Figure 33). SSC had much shallower cavities that were successfully filled with ZrO2 (Figure 34). The results for SSC were very encouraging because good adherence after oxidation exposure was achieved without the use of an oxygen diffusion barrier to prevent oxidation of the substrate.

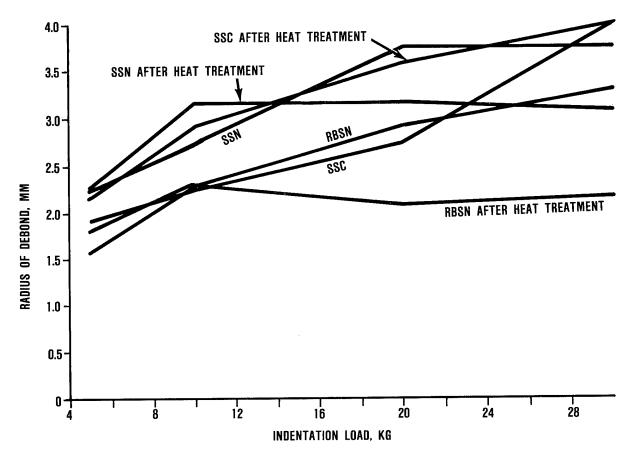


Figure 26. Stoichiometric EB-PVD interlayers resulted in relatively poor initial adherence.

High-Purity Interlayers. ZrO<sub>2</sub>/CVD SiC/SiC specimens were evaluated for coating adherence using indentation and line contact testing. For indentation tests, both as-coated and oxidized specimens spalled at the lowest test loads. Line contact test results were similar as spalling occurred at the lowest contact loads. Because of these preliminary results, high-purity interlayer efforts were not continued.

<u>Diffusion/Gradation Zones</u>. SSN and SSC specimens with nitrogen ion-mixed sputtered Al interlayers were evaluated for ZrO<sub>2</sub> adherence using line contact testing. The RBSN system was not evaluated since cracking and spalling occurred on heat treatment (Figure 35). Contact testing was conducted on specimens with post-coating heat treatments. The results suggested poor adherence for the SSN and SSC systems.

Coating Temperature. As-machined substrates coated at 1040C rather than the baseline 980C were evaluated for adherence using indentation testing. The results showed that coating at elevated temperatures does not appear to benefit the coating adherence before or

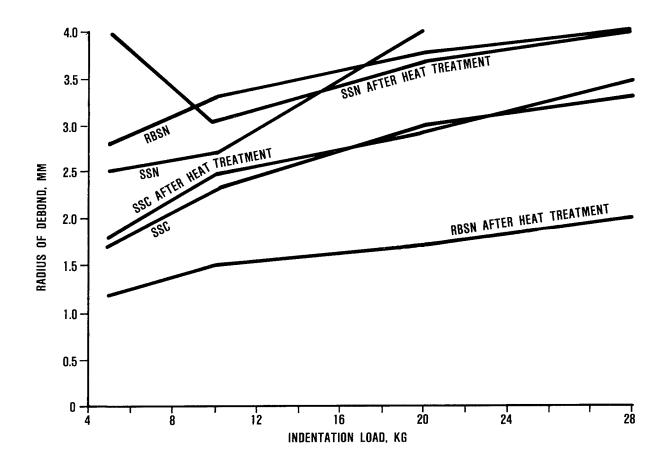


Figure 27. Non-stoichiometric EB-PVD Al<sub>2</sub>O<sub>3</sub> interlayers resulted in poor initial coating adherence. The ZrO<sub>2</sub> adherence was slightly less than observed for stoichiometric EB-PVD Al<sub>2</sub>O<sub>3</sub>.

after oxidation. Adherence for as-coated specimens was equivalent to the baseline for SSN and SSC and slightly less for RBSN. After oxidation exposure, both the coating adherence and the baseline systems degraded significantly.

### 3.2.4 Selection of Coating/Substrate Systems for Tasks 3 and 4

The most adherent coatings were achieved under the Oxygen Diffusion Barrier and Surface Roughening Studies. From these coating/substrate systems, the two most adherent systems for each substrate were selected for analysis under Tasks 3 and 4. Adherence after oxidation exposure was emphasized more than as-coated adherence since oxidizing conditions exist in turbine engine applications. The indentation and line contact adherence results (after oxidation) are summarized in Tables 4 and 5, respectively, for the Oxygen Diffusion Barrier and Surface Roughening Studies and are combined in Table 6. Table 6 ranks the adherence of the pretreatments for each substrate on a scale of 0

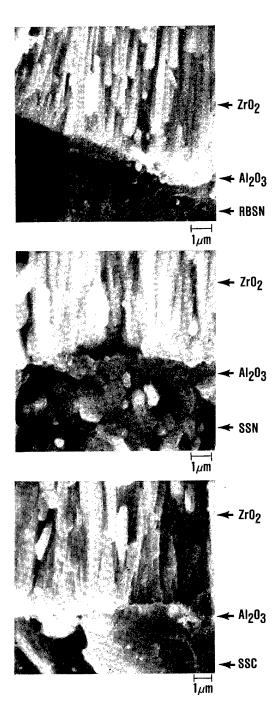


Figure 28. Fracture surfaces of sol-gel alumina interlayer specimens show the coating/interlayer/substrate interfaces exhibiting intimate contact.

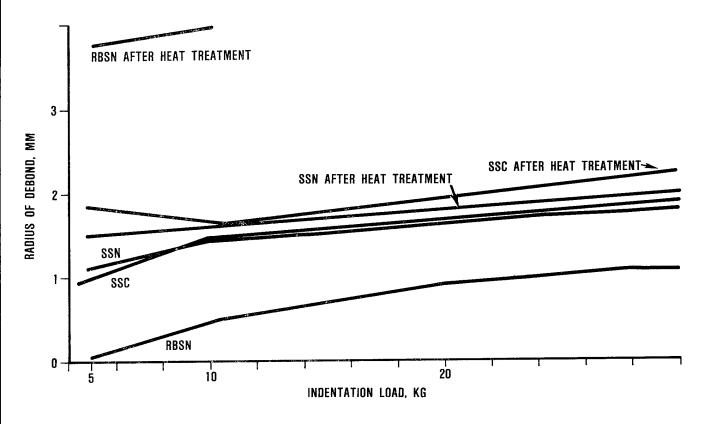


Figure 29. Sol-gel Al<sub>2</sub>O<sub>3</sub> interlayers yielded fair adherence for SSN and SSC which was retained through heat treatment. RBSN exhibited excellent initial adherence but exhibited a large adherence loss on heat treatment.

The results show that difbest. is worst and 6 is to 6; where 0 ferent pretreatments work best for different substrate materials. Al<sub>2</sub>O<sub>3</sub> and CVD AlN interlayers yielded excellent adherence for RBSN. The best adherence for SSN and SSC was achieved using sol-gel alumina interlayers and laser-texturing, though the level of adherence was less than observed for the ZrO<sub>2</sub>/CVD Al<sub>2</sub>O<sub>3</sub>/RBSN and ZrO<sub>2</sub>/CVD Aln/RBSN Laser texturing actually resulted in poor adherence for SSN systems. However, the adherence can probably be improved by (2 ranking). reproducing the surface topography obtained for laser-textured SSC, which had good to excellent adherence (5 ranking). Because the potential for improvement exists, laser-texturing was selected over 150grit machining, which also had an adherence ranking of 2.

## 3.2.5 Task 2 Summary and Conclusions

During Task 2,  $Y_2O_3$  stabilized  $ZrO_2$  coatings were successfully deposited by EB-PVD on substrates having various pretreatments. The  $ZrO_2$  coatings appeared as clean and adherent as those deposited in Task 1.  $ZrO_2$  coatings deposited at a higher temperature (1040C rather

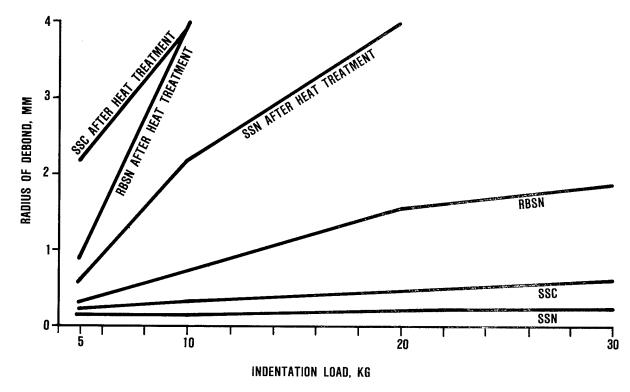


Figure 30. Results for substrates with 150 grit surface finish.

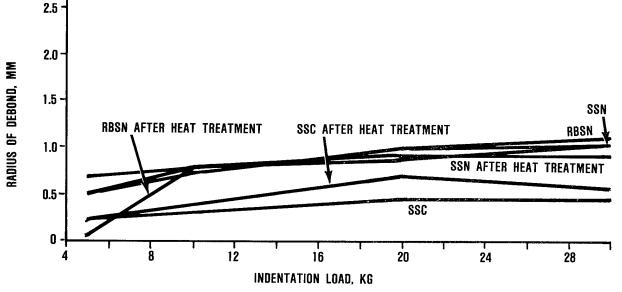


Figure 31. Laser-textured specimens exhibited excellent coating adherence before and after heat treatment. The most encouraging results were observed for laser textured SSC.

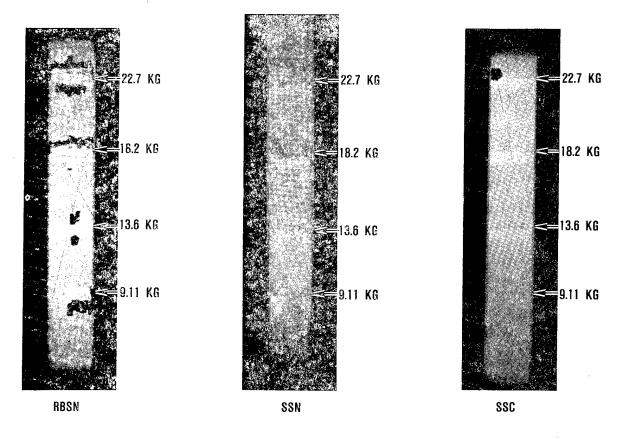


Figure 32. Contact test results on coated laser textured specimens after oxidation suggest poor adherence for RBSN and SSN and good adherence for SSC.

than 980C) also appeared clean and adherent, but did not provide any additional coating adherence. EB-PVD mullite coatings and interlayers could not be applied because the silica in the mullite source evaporated more easily than the alumina resulting in silica-rich coatings.

Oxygen diffusion barriers were developed which successfully prevented adherence degradation previously observed for coated specimens after oxidation. CVD  ${\rm Al}_2{\rm O}_3$  and CVD AlN interlayers worked best for RBSN, yielding excellent coating adherence after oxidation. Sol-gel  ${\rm Al}_2{\rm O}_3$  worked best for SSN and SSC, yielding fair adherence after oxidation.

CVD  ${\rm Al}_2{\rm O}_3$  and CVD AlN interlayers worked well for the porous RBSN substrates but not for the dense SSN and SSC substrates. It is speculated that thermal expansion mismatches between the dense interlayer and the substrate result in interfacial cracking, and that the residual porosity in the RBSN enables islands of interlayer to remain adherent due to rooting in the substrate pores.

Sol-gel  ${\rm Al}_2{\rm O}_3$  interlayers are preferred for SSN and SSC substrates over CVD  ${\rm Al}_2{\rm O}_3$  since relatively good adherence was retained after oxidation, although the level of adherence was less than that observed for  ${\rm ZrO}_2/{\rm CVD}$   ${\rm Al}_2{\rm O}_3/{\rm RBSN}$ . It appears that the 50 percent porous sol-gel interlayer (submicron-sized porosity) provides improved

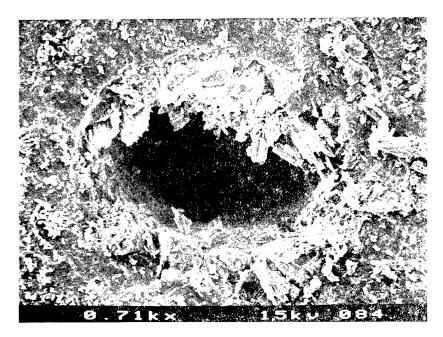


Figure 33. The EB-PVD  $\rm ZrO_2$  did not fill in the deep laser cavities of  $\rm Si_3N_4$  substrates (RBSN shown above) but only coated the cavity walls.

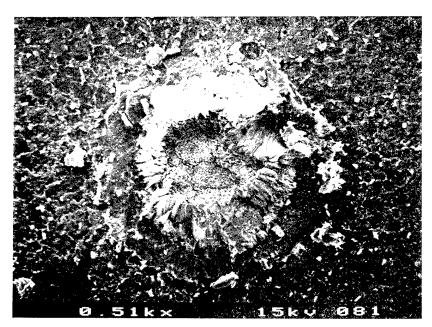


Figure 34. The EB-PVD ZrO<sub>2</sub> infiltrated and filled in the shallow laser cavities on SSC.

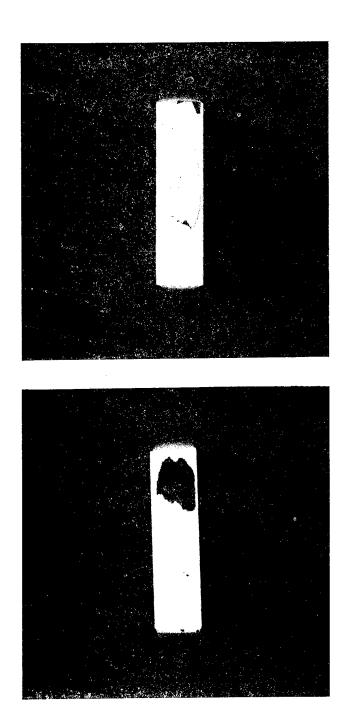


Figure 35. ZrO<sub>2</sub> spalling occurred on heat treatment of RBSN substrates with Ion-mixed (Ar) sputtered Al interlayers.

Table 4. Indentation results (after oxidation) for oxygen diffusion barrier and surface roughening studies

Pretreatment	RBSN	SSN	ssc
CVD Al <sub>2</sub> O <sub>3</sub>	excellent	poor	poor
CVD Aln	excellent	fair	fair
EB-PVD Al <sub>2</sub> O <sub>3</sub> (stoichiometric)	poor	poor	poor
EB-PVD Al <sub>2</sub> O <sub>3</sub> (non-stoichiometric)	poor	poor	poor
Sol-gel Al <sub>2</sub> O <sub>3</sub>	poor	fair	fair
150-grit machine	poor	poor	poor
Laser texture	poor	good	excellent

Table 5. Line contact results (after oxidation) for oxygen diffusion barrier and surface roughening studies

Pretreatment	RBSN	SSN	SSC
CVD Al <sub>2</sub> O <sub>3</sub>	excellent	fair	fair
CVD AIN	excellent**	poor	fair
EB-PVD Al <sub>2</sub> O <sub>3</sub> (stoichiometric)	good*	fair*	good*
EB-PVD Al <sub>2</sub> O <sub>3</sub> (non-stoichiometric)	poor	poor	poor
Sol-gel Al <sub>2</sub> O <sub>3</sub>	good*	good*	good*
150-grit machine	good	good	fair
Laser texture	poor	poor	good

<sup>\*</sup>Lower than typical coefficient of friction.

Table 6. Adherence summary\* (after oxidation) for oxygen diffusion barrier and surface roughening studies

Pretreatment	RBSN	SSN	ssc
CVD Al <sub>2</sub> O <sub>3</sub>	6	1	1
CVD AlN	6	1	2
EB-PVD Al <sub>2</sub> O <sub>3</sub> (stoichiometric)	2	1	2
EB-PVD Al <sub>2</sub> O <sub>3</sub> (non-stoichiometric)	0	0	0
Sol-gel Al <sub>2</sub> O <sub>3</sub>	2	3	3
150-grit machine	2	2	1
Laser texture	2	2**	5

<sup>\*</sup>Adherence rankings assigned to pretreatment/substrate combinations by giving numerical values to qualitative results in Tables 4 and 5 (poor = 0, fair = 1, good = 2, and excellent = 3), then adding the tables together (0 = worst, 6 = best).

<sup>\*\*</sup>Not tested in contact rig; manually scraped.

<sup>\*\*</sup>Chosen over 150-grit machining because potential for improvement exists.

strain tolerance, which limits interfacial cracking due to thermal expansion mismatches but at the cost of interlayer coehesive strength.

Roughening the substrate through laser-texturing significantly improved both the as-coated and oxidized coating adherence for SSC. The good adherence observed for the laser-textured SSC after oxidation exposure was encouraging considering no interlayer was used to prevent oxidation of the interface. Results were not as encouraging for laser-textured RBSN and SSN. However, the surface cavities were much deeper than those produced on SSC and were not completely filled in with  $ZrO_2$ . Therefore, the potential for improvement exists. One negative aspect of laser-texturing is that the coating adherence improvement is accompanied by a decrease in substrate strength associated with the surface cavities.

The two pretreatments yielding the best  ${\rm ZrO_2}$  coating adherence for each substrate were selected for analysis in Tasks 3 and 4. The most adherent systems were chosen based on adherence after oxidation (Table 7). Different pretreatments worked best for different substrate materials. CVD  ${\rm Al}_2{\rm O}_3$  and CVD AlN interlayers yielded the best adherence for RBSN, while sol-gel alumina interlayers and laser-texturing gave the best adherence for SSN and SSC.

### 3.3 TASK 3 - CONTACT STRESS TESTING AND FRICTION MEASUREMENTS

In Task 3, the most adherent coating/substrate systems from Task 2 were evaluated for strength, resistance to thermal spalling, and friction characteristics. Friction test specimens were exposed to high contact stresses during testing. These same specimens were then used in Task 4, Post-Contact Strength Measurements, to evaluate contact stress damage resistance.

### 3.3.1 Specimen Preparation and Inspection

Task 3 substrate pretreatments, selected at the conclusion of Task 2, are shown in Table 7. Test bars of each substrate material were procured, pretreated and EB-PVD coated according to the procedures used in Task 2. The laser texturing procedures were modified to yield shallower surface cavities in an attempt to minimize the strength loss associated with this treatment. Also, shallower cavities should yield better coating adherence for SSN than achieved in

Table 7. Pretreatments selected for Tasks 3 and 4

RBSN	SSN	ssc
CVD alumina CVD aluminum nitride	sol-gel alumina laser machining	sol-gel alumina laser machining

Task 2.\* Extra laser-textured substrates were prepared for strength evaluation. In addition, as-machined substrates were prepared to provide baseline strength, friction, and contact damage resistance. All EB-PVD zirconia-coated specimens designated for testing were heat-treated at 1200C for 100 hours in order to stabilize the interface prior to evaluation. A few as-coated specimens were used for interface analysis.

All coated specimens were visually inspected after EB-PVD coating. Most coatings appeared clean and adherent, although coating discoloration was observed for coatings from EB-PVD Run No. 124, which involved CVD  $\rm Al_2O_3$ -coated RBSN. Most of the specimens from this run were brown, some having a mottled appearance. However, EDX analysis of the zirconia coating surface detected no chemical impurities. On heat treatment, several of these specimens spalled and most exhibited a rust-colored reaction product on the substrate. This effect was not observed for specimens from any other coating run. EDX analysis of a spalled area suggested that the  $\rm Al_2O_3/RBSN$  substrate was contaminated prior to EB-PVD coating; EDX detected potassium (Figure 36). The strong aluminum peak suggests that the failure occurred at the  $\rm ZrO_2/Al_2O_3$  interface.

For specimens other than those from Run No. 124, only color changes in the  $\rm ZrO_2$  coating from blue or gray to white (due to oxidation of excess zirconium in the  $\rm ZrO_2$  coating) occurred on heat treatment.

#### 3.3.2 Flexure Strength

Flexure strength testing was conducted on both the as-machined substrates and coated specimens. The as-machined substrate strengths served as a baseline in determining the effect of the coating procedures on the substrate strength. The coated strengths served as a baseline in determining the retained strength (or contact damage resistance) after contact stress exposure.

3.3.2.1 Flexure Test Procedures. Flexure testing was conducted at room temperature, 1000C, and 1375C in air. The procedures for room temperature testing were the same as those used in Task 1. Similar procedures were used at elevated temperatures except that specimens were held for 5 minutes at temperature prior to starting each test. The flexure strength test matrix is shown in Table 8.

Because Task 2 flexure results identified a strength loss associated with laser-texturing, additional testing of uncoated, laser-textured SSN and SSC was conducted in order to separate the strength effects of laser-texturing and coating.

<sup>\*</sup>During Task 2, poor EB-PVD ZrO<sub>2</sub> coating adherence resulted from incomplete filling of the deep surface cavities in laser-textured SSN.

# ENERGY DISPERSIVE X-RAY SPECTRUM

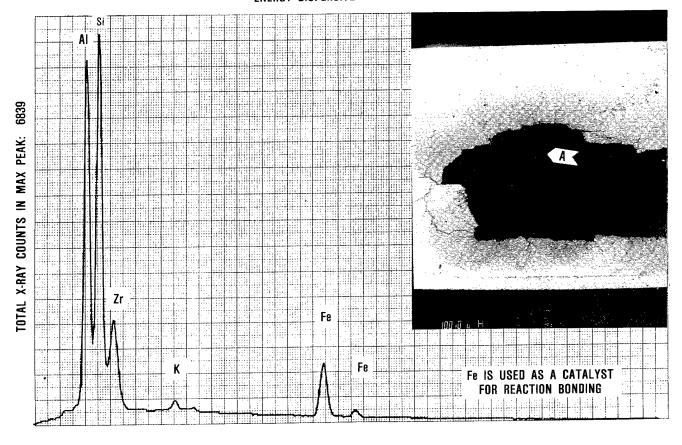


Figure 36.  ${\rm ZrO}_2$  coatings from run #124 spalled during heat treatment. Potassium (K) was detected in the spalled area.

Table 8. Task 3 flexure strength test matrix

	N	Number of tests		
Substrate	25C	1000C	13750	
RBSN	10	6	6	
SSN	10	6	6	
SSC	10	6	6	
Coating/substrate system		_		
Coating/substrate system ZrO2/CVD Al2O3/RBSN	10	6	6	
ZrO2/CVD Al2O3/RBSN	10 10	6 6	6	
ZrO <sub>2</sub> /CVD Al <sub>2</sub> O <sub>3</sub> /RBSN ZrO <sub>2</sub> /CVD Aln/RBSN		_	-	
ZrO <sub>2</sub> /CVD Al <sub>2</sub> O <sub>3</sub> /RBSN ZrO <sub>2</sub> /CVD Aln/RBSN ZrO <sub>2</sub> /sol-gel Al <sub>2</sub> O <sub>3</sub> /SSN	10	6	6	
ZrO <sub>2</sub> /CVD Al <sub>2</sub> O <sub>3</sub> /RBSN ZrO <sub>2</sub> /CVD Aln/RBSN	10 10	6 6	6 6	

After flexure testing, all fracture surfaces were examined optically up to 40% magnification to determine the failure origins for each system. The fractography results of as-machined substrates and coated specimens were compared to determine if the coating procedures had any effect on the failure characteristics.

3.3.2.2 Flexure Test Results. The flexure test results for the as-machined substrates are shown in Table 9. The strengths measured for RBSN and SSN are typical for these materials based on past GTEC experience. SSC strengths were lower than typical (340 to 400 MPa). In most cases, RBSN failed from either residual porosity or light SSN exhibited colored inclusions at room and elevated temperatures. high-energy failures at room temperature (bars broke into several pieces). The origins were primarily metallic-looking inclusions. 1000C, low-energy fractures occurred (bar broke in two pieces having flat, relatively smooth fracture surfaces). All failures originated from the tensile face. At 1375C, SSN exhibited creep deformation as evidenced by the nonlinear load deflection curves and a rough fracture surface. SSC exhibited surface failures at room temperature and 1000C. At 1375C, the origins were light colored areas, usually associated with residual porosity in SSC.

Table 9. As-machined substrate strength

Substrate	Temperature (C)	Strength ± std dev (MPa)
RBSN	RT*	296 ±31
	1000	361 ±18
	1375	319 ±52
SSN	RT	592 ±67
	1000	611 ±51
	1375	134 ± 6
SSC	RT	286 ±26
	1000	272 ±21
	1375	295 ±12

\*Room Temperature

The flexure test results for laser-textured SSN and SSC (Table 10) show that a significant strength loss results from laser texturing. Laser-textured specimens failed at the cavities produced by laser-texturing except for SSN at 1375C, where softening of the grain boundary phases predominated. For SSC, several of the failure origins were associated with spalling on the chamfered edges of the test bars which occurred during laser-texturing (Figure 37). To evaluate the effect on strength, some SSC specimens were laser-machined over the

Substrate	Temperature (C)	Strength ± std dev (MPa)	Reduction (percent)
SSN	RT	450 ±30	24
	1000	502 ±57	18
	1375	111 ± 7	17
SSC	RT	205 ±17	28
	1000	224 ± 8	18
	1375	225 ±13	24

Table 10. Laser-textured substrate strength

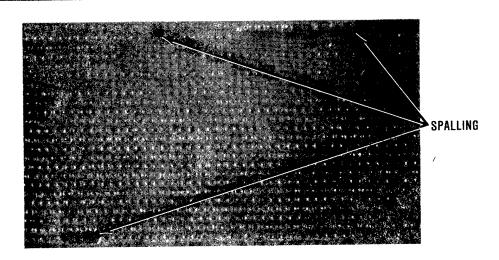


Figure 37. Laser machining of SSC substrates resulted in spalling on edges.

center section of the bar, leaving the edges non-textured by the laser (Figure 38). As shown in Table 11, the strength reduction was not as significant. Afterwards, all laser-textured SSC specimens to be coated were rechamfered on the edges to eliminate the effects of edge spalling. Therefore, the results in Table 11 are expected to better represent the strength characteristics of laser-textured SSC.

The room temperature strengths of the Task 3 laser-textured specimens were lower than exhibited by Task 2 laser-textured specimens which had cavity depths up to three times greater. This result suggests that the strength was not predominated by the cavity depth but possibly by microcracks extending from the cavities. Both SSN and SSC exhibited less strength reduction when tested at elevated temperatures. Possible explanations are that microcracks extending from the laser cavities blunt or heal during the test (5 minutes hold at temperature) or that laser-texturing produces residual stress effects relieved at elevated temperatures.

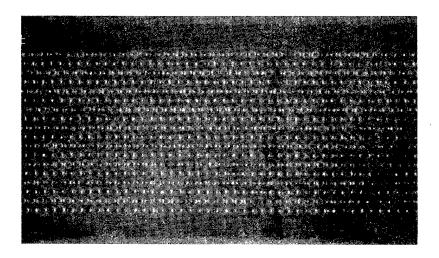


Figure 38. Some specimens were left untreated at the edges to eliminate spalling.

Table 11	. Strengt	th of	rechamfered	laser-textured	SSC

Substrate	Temperature (C)	Strength ± std dev (MPa)	Reduction (percent)
SSC	RT	220 ±10	23
	1000	$241 \pm 3$	11
	1375	271 ±13	8

The flexure test results for the six coated systems are summarized in Table 12. Both the  $\rm ZrO_2/CVD$   $\rm Al_2O_3/RBSN$  and  $\rm ZrO_2/CVD$   $\rm AlN/RBSN$  systems exhibited significantly lower room temprature strength than as-machined RBSN (-29 and -21 percent, respectively). However, these two systems regained strength at elevated temperatures, suggesting that the strength loss at room temperature was a residual stress effect. At 1375C the  $\rm ZrO_2/CVD$   $\rm Al_2O_3/RBSN$  and  $\rm ZrO_2/CVD$   $\rm Aln/RBSN$  systems exhibited strengths higher than baseline RBSN (+10 and +4 percent, respectively). Fractography results suggested that the room temperature and 1000C failures originate from substrate surface flaws and the 1375C failures from internal flaws. SEM analysis of these specimens showed that both the surface and internal origins were associated with the residual porosity of the RBSN (Figure 39).

The  $\rm ZrO_2/sol-gel~Al_2O_3/SSN$  system exhibited approximately 35 percent lower strength at room temperature and 1000C relative to asmachined SSN. Fractography suggested that the strength loss was not associated with the coating since the failures originated at large, yttria-based (sintering additive in SSN) flaws, as shown in Figure 40. The cause of the strength loss was determined to be due to the heat treatment at 1200C for 100 hours. Fifteen as-machined SSN substrates were exposed to the same heat treatment to support this hypothesis. A strength of 435  $\pm$ 71 MPa resulted, a 27 percent drop. Therefore, the

Table 12. Baseline flexure strength results for coated systems (heat-treated at 1200C for 100 hours)

Coating/substrate system	Temperature (C)	Strength ± std dev (MPa)
ZrO <sub>2</sub> /CVD Al <sub>2</sub> O <sub>3</sub> /RBSN	RT	211 ±33
2. 2 J.	1000	303 ±32
	1375	352 ±49
ZrO2/CVD Aln/RBSN	RT	234 ±39
2	1000	321 ±59
	1375	333 ±78
ZrO <sub>2</sub> /sol-gel Al <sub>2</sub> O <sub>3</sub> /SSN	RT	383 ±46
2 - 23	1000	411 ±19
	1375	125 ±17
ZrO <sub>2</sub> /laser-textured SSN	RT	353 ± 9
-	1000	408 ±101
	1375	125 ±26
ZrO <sub>2</sub> /sol-gel Al <sub>2</sub> O <sub>3</sub> /SSC	RT	279 ±23
2 - 2 3	1000	363 ±22
	1375	268 ±32
ZrO <sub>2</sub> /laser-textured SSC	RT	238 ±46
-	1000	288 ±22
	1375	263 ±31

lower strength observed for  $\rm ZrO_2/sol-gel~Al_2O_3/SSN$  appears to be independent of the coating, and 435 MPa was subsequently used as the room temperature and 1000C baseline for SSN.

The  $\rm ZrO_2/sol-gel~Al_2O_3/SSC$  system exhibited no change in strength at room temperature and 1375C relative to the as-machined SSC.\* At 1000C, a 34 percent increase in strength was observed. However, since all failures still originated from residual porosity in the SSC, the strength increase at 1000C is believed to be coincidental.

Both the  $\rm ZrO_2/laser$ -textured SSN and  $\rm ZrO_2/laser$ -textured SSC systems exhibited some strength loss at room temperature relative to their respective substrate materials.\*\* At elevated temperatures, the strength

<sup>\*</sup>The baseline strengths observed for SSC under the current program were less than typical for this material. The lower strength was attributed to the atypical residual porosity observed.

<sup>\*\*435</sup> MPa was used as the baseline for SSN because of the effect of the heat treatment on strength.

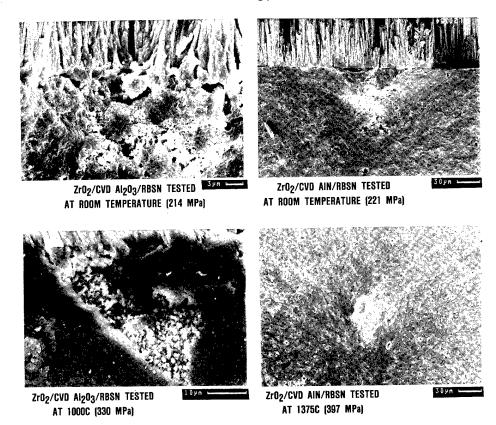


Figure 39. Both surface and internal failures for the  $\rm ZrO_2/CVD$   $\rm Al_{2}O_{3}/RBSN$  and  $\rm ZrO_2/CVD$   $\rm Aln/RBSN$  systems are associated with residual porosity in RBSN.

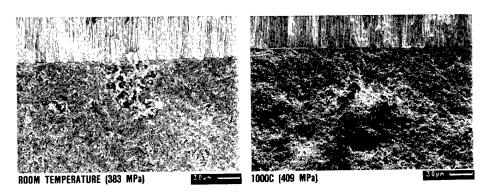
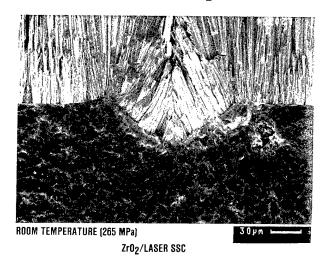


Figure 40. ZrO<sub>2</sub>/Sol-gel Al<sub>2</sub>O<sub>3</sub>/SSN substrates failed primarily from large substrate flaws.

changes were insignificant ( $\rm ZrO_2/laser$ -textured SSC actually exhibited a higher strength at 1000C than SSC). The failure origins were at the surface cavities produced by laser-texturing (except for  $\rm ZrO_2/laser$ -textured SSN at 1375C where deformation in fast fracture was observed), as shown in Figure 41.

For all coating/substrate systems, glass buildup occurred on the  $\rm ZrO_2$  coatings under the hot-pressed SiC flexure load pins at 1375C (Figure 42). The glass buildup was most significant for systems exhibiting the highest strength at 1375C (e.g.,  $\rm ZrO_2/CVD~Al_2O_3/RBSN$  and  $\rm ZrO_2/CVD~AlN/RBSN$ ). Therefore, the magnitude of the load on the pin is a factor. These results suggest that there may be some sticking problems during the 1375C coated contact testing (SSC contact pins are to be used against the  $\rm ZrO_2$  coatings). To further assess this problem, studies



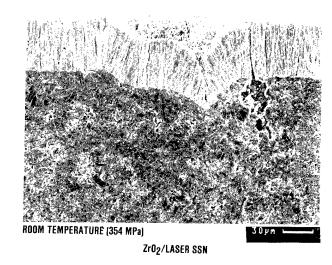


Figure 41. Failures in laser-textured specimens initiated at surface cavities on substrates.

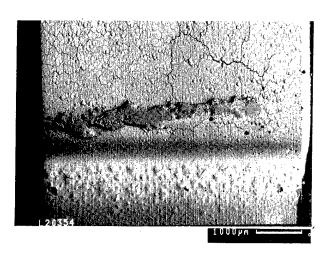


Figure 42. The EB-PVD  ${\rm ZrO_2}$  coatings exhibited glass buildup under the flexural load pins during 1375C tests.

were conducted to evaluate the sticking resistance of EB-PVD  $\rm ZrO_2$  against itself and the three substrate materials. The results indicate the  $\rm ZrO_2$  coating sticks to  $\rm SiO_2$  formers (e.g.,  $\rm Si_3N_4$  and  $\rm SiC$ ) but not to itself. Therefore, in heat engine applications at this temperature, both contacting surfaces would need to be coated to prevent sticking.

### 3.3.3 Cyclic Thermal Testing

3.3.3.1 Cyclic Thermal Test Procedures. Thermal cyclic testing was conducted in GTEC's cyclic thermal shock rig. The Task 3 coating/substrate systems were tested according to the test matrix in Table 13.

Each test consisted of three cycles. The specimens were heated to the test temperature in 15 seconds or less then airblast cooled to less than 150C. This cycle was repeated two times, followed by visual inspection of the specimens to check for spalling delamination or coating degradation.

3.3.3.2 Cyclic Thermal Test Results. The results for 1000C and 1375C cyclic thermal tests are shown in Figures 43 and 44, respectively. Three systems survived cyclic testing to 1000C: ZrO2/CVD Al2O3/RBSN, ZrO2/CVD AlN/RBSN, and ZrO2/sol-gel Al2O3/SSN. These systems exhibited no visual evidence of reaction or degradation. For the three remaining systems, spalling occurred only on the chamfered edges where localized heating was observed during testing. For the ZrO2/laser-textured SSN and ZrO2/laser-textured SSC systems, only one of the four specimens tested exhibited edge spalling. Two ZrO2/sol-gel Al2O3/SSC specimens edge spalled.

Three systems survived cyclic testing to 1375C: ZrO<sub>2</sub>/CVD Al<sub>2</sub>O<sub>3</sub>/RBSN, ZrO<sub>2</sub>/laser-textured SSN, and ZrO<sub>2</sub>/laser-textured SSC. These systems exhibited no visual evidence of reaction or degradation. For ZrO<sub>2</sub>/CVD AlN/RBSN, one of four specimens exhibited spalling during the first quench. No further spalling occurred during subsequent cycles. The other specimens survived all three cycles. ZrO<sub>2</sub>/sol-gel Al<sub>2</sub>O<sub>3</sub>/SSN and SSC exhibited poor spalling resistance in 1375C tests.

Table 13. Cyclic thermal test matrix and results

Number of						
Coating/substrate	te	sts		Results		
system	1000C	1375C		1000C		1375C
ZrO2/CVD Al2O3/RBSN	4	4		OK		OK
ZrO2/CVD Aln/RBSN	4	4		OK	1	spalled
ZrO2/sol-gel Al2O3/SSN	4	4		OK	4	spalled
ZrO2/laser-textured SSN	4	4	1	spalled+		OK
ZrO2/sol-gel Al2O3/SSC	4	4	2	spalled	3	spalled
ZrO2/laser-textured SSC	4	4	1	spalled+		OK

<sup>+</sup> minor edge spalling only

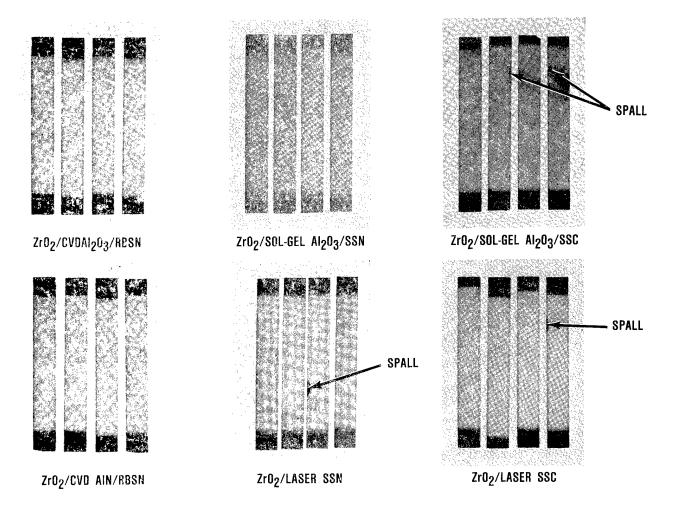


Figure 43. Three systems survived 1000C cyclic thermal testing while the remaining systems exhibited some edge spalling.

### 3.3.4 Uncoated Contact Testing

Contact testing was conducted on as-machined RBSN, SSN, and SSC to establish the baseline friction characteristics and, through subsequent flexure strength testing, to determine the threshold contact test conditions for damaging the substrate materials. The threshold contact test conditions established were used to select test conditions for coated contact tests.

3.3.4.1 Uncoated Contact Test Procedures. Contact tests were conducted using both point and line contact geometries. The original test plan included line contact only since prior analyses of Si3N4 and SiC ceramics suggested the test conditions selected should induce contact damage. However, problems were encountered in obtaining thresholds for contact damage. Therefore, point contact, which generates higher contact stresses, was incorporated into the test plan.

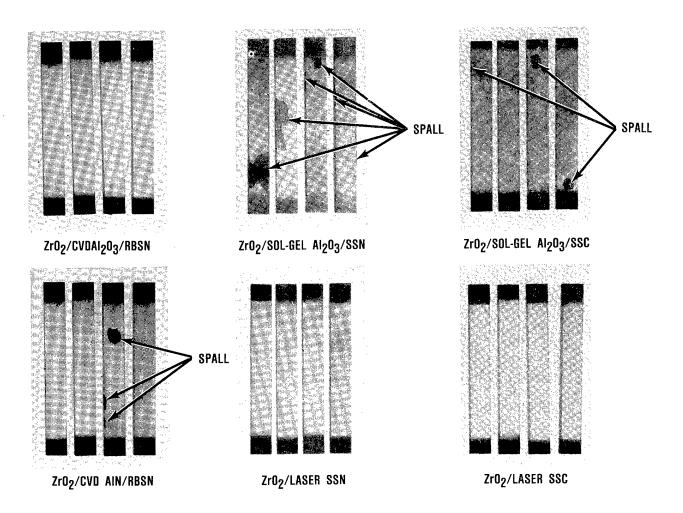


Figure 44. Three systems survived 1375C cyclic testing. The remaining systems exhibited face and edge spalling.

Contact testing involved displacing a normal force, applied through a SSC contact pin onto the test specimen at its mid-span, 1.52 mm across the surface while monitoring the tangential force (see Section 3.1.3.2 for description of test rig). The original test matrix for contact testing, which included only line contact, is shown in Table 14. Tests were conducted up to 1375C for RBSN and SSC. SSN was only tested to 1200C since flexure test results suggested 1375C was beyond its range of use. For elevated temperature tests, the normal load was applied and held 30 minutes at temperature before displacing the contact. This hold period allowed for sticking at the contact interface due to substrate oxidation. Sticking of component interfaces is frequently observed in ceramic gas turbine engines.

Because line contact did not yield the contact damage threshold information necessary to choose the test conditions for coated contact specimens, more severe test conditions were pursued. Contact testing

Table 14. Line contact test matrix

Temperature (C)	4.5	Load (kg) 11.3	22.7
RT	4	4	4
1000	4	4	4
1375*	4	4	4

\*SSN tested at 1200C instead of 1375C

was conducted on spare substrate specimens using a point contact geometry. As a first step, tests were conducted at the same temperatures and loads as used for line contact. For SSN\*, testing was extended to loads of 27.2, 39.3, and 45.4 kg. The limited number of substrates available resulted in the test matrix shown in Table 15.

Contact tested specimens were flexure tested at room temperature to determine the retained strength. The contact tested side of the specimen was in tension. All fracture surfaces were examined up to 40% to characterize the failure origins as either contact or non-contact.

## 3.3.4.2 Uncoated Contact Test Results.

Friction Characteristics. The static and dynamic friction characteristics observed for the as-machined substrate materials are

Table 15. Point contact test matrix

Substrate	Temperature (C)	Load (kg)					
		4.5	11.3	22.7		39.3	45.4
RBSN	RT		1	2			
	1000			2			
	1375			2			
SSN	RT	3	3	3	3	4	4
	1000	3	3	3			
	1200	3	3	3			
SSC	RT	3	3	3			
	1000		1	3			
	1375		1	3			

<sup>\*</sup>The SSN substrates were heat treated 100 hours at 1200C to provide a better baseline for coated SSN contact results.

summarized in Figures 45 and 46, respectively. The substrate materials exhibited significantly higher static coefficients of friction at elevated temperature. This trend was also observed for the dynamic friction of SSN and SSC. For RBSN, the increase was not as significant. The increased friction coefficient at elevated temperatures was attributed to SiO<sub>2</sub> formation on the substrate materials in oxidizing atmospheres. Based on GTEC's prior experience with Si<sub>3</sub>N<sub>4</sub> and SiC ceramics, the friction coefficients observed are characteristic of the materials evaluated. 7,8

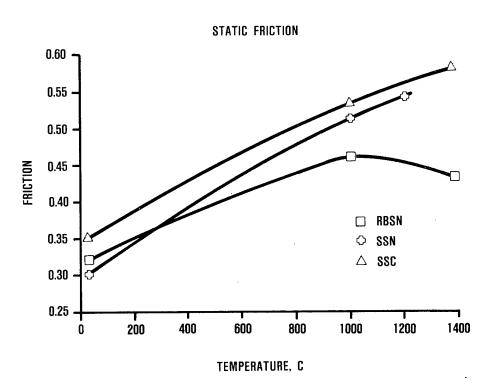


Figure 45. The static friction of RBSN, SSN, and SSC increased at elevated temperatures.

Retained Strength. Strength results indicated line contact testing up to 22.7 kg was inadequate to cause contact stress damage in most cases. The only contact damage observed was for RBSN; two specimens exhibited 25 percent reductions in strength after contact testing at 1000C and 1375C using 22.7 kg normal loads. SSN exhibited a strength loss compared to baseline after 1200C testing. However, the strength loss was due attributed to the effect of elevated temperature exposures on SSN discussed in Section 3.3.2.2 since no failures initiated from the line of contact. SSC did not exhibit any strength losses. Prior GTEC experience suggests the line contact line loads utilized should have exceeded the thresholds for damaging RBSN and SSC (no data base existed for SSN). However, all prior contact test experience with materials at

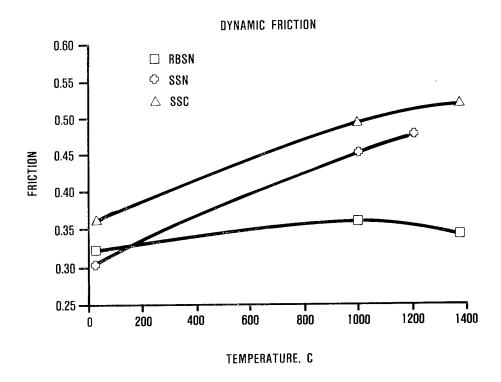


Figure 46. The dynamic friction of RBSN, SSN, and SSC increased at elevated temperatures. The increase for RBSN was minor.

the loads of interest involved the use of a non-self-aligning contact test rig rather than the self-aligning rig used in the current program. The non-self-aligning system may have produced a concentrated contact load which is avoided in the self-aligning rig.

Point contact testing provided the higher contact stresses required to develop thresholds for RBSN and SSC at room and elevated temperatures. Threshold conditions were obtained for SSN at room temperature, but sufficient specimens were not available for obtaining elevated temperature thresholds. The results of the point contact tests are summarized in Table 16.

RBSN exhibited contact damage for all point contact tests.

SSN did not exhibit any contact damage up to 22.7 kg normal load. Therefore, the load range was extended at room temperature. At 27.2 kg, one of the three specimens tested exhibited contact damage and a strength loss of 40 percent. At 39.3 kg load, all specimens exhibited an average strength loss of 45 percent. At 45.4 kg, problems were encountered in testing. The end of the bottom load rod of the test rig contacted the upper load rod and shared some of the contact load applied. However, a 30 percent strength loss due to contact stress damage was still observed. Due to a limited quantity of SSN material, specimens were not available for additional elevated temperature testing at higher loads.

					oad kg)		
Substrate	Temperature (C)	4.5	11.3	22.7	27.2	39.3	45.4
RBSN	RT		-35%	-40%			
	1000			-45%			
	1375			-30%			
SSN	RT	ND+	ND	ND	-40%	-45%	-45% -30%+
	1000	ND	ND	ND			
	1200	ND	ND	ND			
SSC	RT	ND	-12%	-30%			
	1000		ND	-30%			
	1375		ND	-35%			

Table 16. Point contact test results\*

SSC exhibited no contact damage for 4.5 kg load tests at room temperature. At 11.3 kg, one of three specimens was damaged and the strength loss was 12 percent. At 22.7 kg, all specimens exhibited contact damage at room temperature with an average strength loss of 30 percent. At 1000C, 22.7 kg loading caused damage on two of three SSC specimens; the average strength loss was 30 percent. Only one of three specimens exhibited contact damage at 1375C and 22.7 kg loading. The strength loss was 35 percent.

# 3.3.5 Coated Contact Testing

3.3.5.1 Coated Contact Test Procedures. Contact testing was conducted on coated specimens using procedures similar to those discussed in Section 3.3.4.1. The test matrices for coated contact testing are shown in Table 17. Preliminary contact tests (Table 17a) were conducted using both point and line contact. Three specimens from each system were evaluated. One specimen was used at each of the test temperatures and four tests were conducted per specimen. The preliminary test matrix provided friction characteristics under the same test conditions for all six coating/substrate systems.

The contact stress tests (Table 17b) were conducted using test conditions above the threshold for damaging the as-machined substrates (determined in uncoated contact tests). Therefore systems with different substrates were tested under different conditions. The contact stress test matrix utilized one specimen per test so that the coated

<sup>\*</sup>Numbers shown are average strength losses for specimens exhibiting contact damage.

<sup>+</sup>Not damaged; no contact damage was exhibited.

<sup>++</sup>Problem in testing at 45.4 kg resulted in a lower normal load on the specimen than applied.

Table 17. Test matrix for coated contact testing

### a. Preliminary friction tests

Temperature	Contact load Line (kg)				
(C)	22.7	4.5	11.3	22.7	
RT	1	1	1	1	
1000	1	1	1	1	
1375*	1	1	1	1	

\*SSN was tested at 1200C instead of 1375C.

## b. Contact stress tests (point contact only)

		Contact load		
	Temperature		(kg)	
Coating/Substrate System	(C)	22.7	27.2	34.3
ZrO2/CVD Al2O3/RBSN	RT	10		
and	1000	10		
ZRO2/CVD Al2O3/RBSN	1375	5	5	
ZrO2/sol-gel Al2O3/SSN	RT			10
and	1000			10
ZrO <sub>2</sub> /laser-textured SSN	1200			10
ZrO2/sol-gel Al2O3/SSN	RT	10		
and	1000	5	5	
ZrO2/laser-textured SSN	1375	5	5	

specimens could be subsequently flexure tested for retained strength (Task 4).

3.3.5.2 Coated Contact Test Results. The coefficients of friction observed for the preliminary friction tests and contact stress tests are shown in Figures 47 and 48, respectively. The friction results from both test matrices agree. The static and dynamic friction coefficients were unexpectedly high. Previous studies on cubic stabilized zirconia (CSZ) and transformation toughened zirconia (TTZ) materials resulted in static and dynamic coefficients of friction less than 0.2 up to 1375C. 9,10 The high modulus SSC contact pin (e.g., 400 MPa) may deform the low modulus coating (e.g., 10 to 20 MPa) to create a plowing effect, which would increase the apparent room temperature friction.

The friction coefficients increased at elevated temperatures which is also uncharacteristic of zirconia. However, the contact pins used

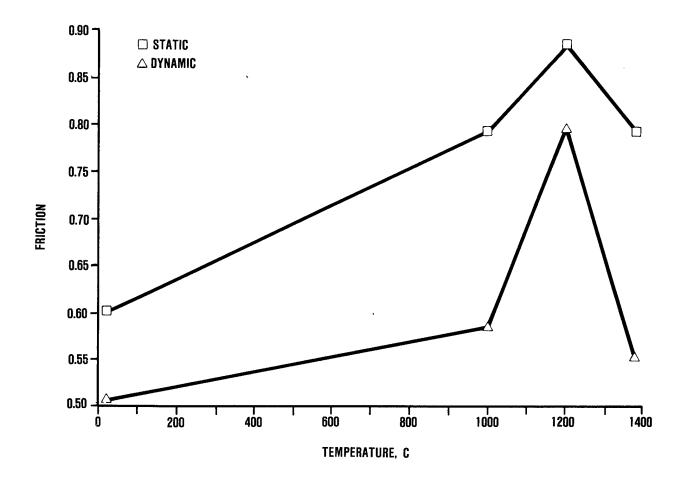


Figure 47. The EB-PVD ZrO<sub>2</sub> coatings exhibited high coefficients of friction during preliminary friction tests.

were SSC, a silica former, whereas the early work on CSZ and TTZ used contact pins of like material. Optical analysis of the contact region showed the coating to be smeared and glassy after 1000C contact exposure. As-discussed in Section 3.3, glass formation on the SSC contact pin was anticipated as a potential problem since glass build-up was observed on the ZrO<sub>2</sub> coating under the flexure load pins after 1375C testing. Since coating failure occurred in all 1375C contact tests, glass build up on the ZrO<sub>2</sub> coating could not be characterized. However, glass formation may have contributed to frequent spalling observed at 1375C. Using a zirconia coated contact pin against the zirconia coated specimen would probably yield more stable friction characteristics.

In many cases, the coating chipped or spalled during contact testing, but the coating friction coefficients could be extracted since the spalling and chipping usually occurred after the point contact had traversed a significant distrance. Some spalling occurred prior to the initiation of sliding, and the static coefficient of friction measured was actually the result of this spalling. After spalling, the dynamic friction coefficient was for the SSC contact pin/substrate interface

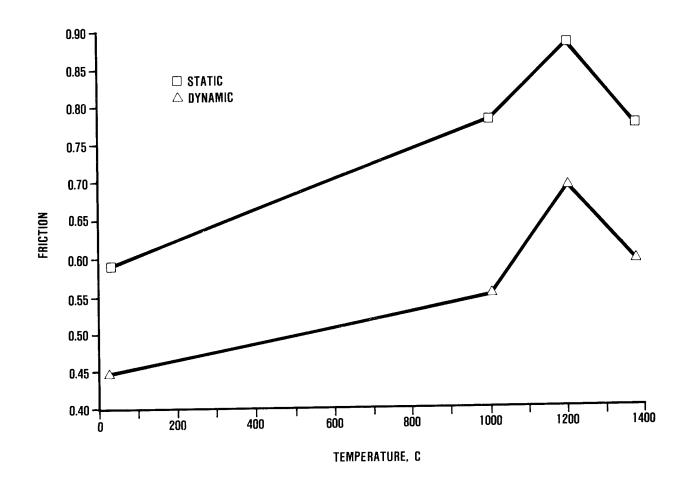


Figure 48. Friction results from contact stress tests were in agreement with the preliminary friction test results.

since the  ${\rm ZrO_2}$  was no longer present. It should be pointed out that the point contact conditions used in these contact stress tests are much more severe than the line contact tests used to assess coating adherence in previous tasks.

At room temperature, most systems exhibited chipping of the zirconia coating in the test zone. Laser-textured SSC performed the best at room temperature (two specimens chipped) followed by CVD Al<sub>2</sub>O<sub>3</sub>/RBSN (four chipped). The chipped areas of the laser-textured SSC still had EB-PVD ZrO<sub>2</sub> in the surface cavities (Figure 49). Both the CVD Aln/RBSN and laser-textured SSN had only a few survivors. Specimens from the ZrO<sub>2</sub>/sol-gel Al<sub>2</sub>O<sub>3</sub>/SSN and ZrO<sub>2</sub>/sol-gel Al<sub>2</sub>O<sub>3</sub>/SSC systems all spalled. The ZrO<sub>2</sub>/sol-gel Al<sub>2</sub>O<sub>3</sub>/SSN system was the only system which exhibited large area spalls. Laser-textured SSN, like laser-textured SSC, had ZrO<sub>2</sub> still adhered in the surface cavities of spalled areas (Figure 50).

At 1000C most specimens exhibited better coating adherence. CVD  $\rm Al_{2}O_{3}/RBSN$ , CVD  $\rm Aln/RBSN$ , and laser-textured SSC, no  $\rm ZrO_{2}$  spalling was exhibited. Only two specimens each spalled for sol-gel  $\rm Al_{2}O_{3}/SSN$  and

# TEST DIRECTION 100.0 U 14 11.1.20.0.34 120 183

Figure 49. For laser-textured SSC, the EB-PVD ZrO<sub>2</sub> remained adherent adherent in the surface cavities even though the coating spalled on contact testing.

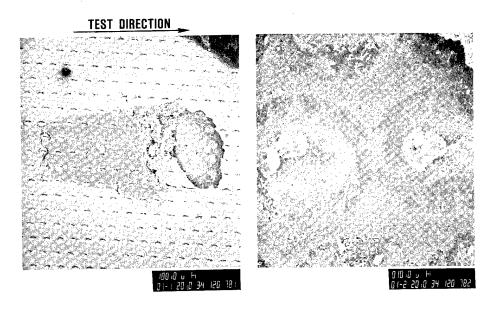


Figure 50. Like for laser-textured SSC, the EB-PVD ZrO<sub>2</sub> remained adherent in the surface cavities of the laser-textured SSN.

SSC. Laser-textured SSN spalled the most frequently (seven of ten spalled). The overall better adherence observed at 1000C may be due to relieving residual stresses at the interface near processing temperature (EB-PVD  $\rm ZrO_2$ , CVD  $\rm Al_2O_3$ , and CVD AlN were deposited near 1000C, and the sol-gel  $\rm Al_2O_3$  were calcined at 1000C).

Very few systems had specimens whose coating survived during 1200C or 1375C contact tests. The only survivors were three laser-textured SSN specimens tested at 1200C using 34.3 kg normal load.

# 3.3.6 Task 3 Summary and Conclusions

Strength Testing. The coating/substrate systems were flexure tested to determine the effect of pretreatment, coating, and heat treatment on substrate strength.  $\rm ZrO_2/CVD~Al_2O_3/RBSN$  and  $\rm ZrO_2/CVD~AlN/RBSN$  showed a strength loss at room temperature but regained the lost strength at elevated temperature. At 1375C, both systems exhibited higher than baseline strength. The results suggest that residual stresses may be effecting the strength of these systems.

ZrO<sub>2</sub>/sol-gel Al<sub>2</sub>O<sub>3</sub>/SSN and ZrO<sub>2</sub>/laser-textured SSN both exhibited strengths at room temperature and 1000C much lower than as-machined SSN. However, the strength loss was primarily due to the degradation of the SSN substrate during heat treatment since uncoated SSN exhibited a 27 percent strength loss after an exposure 1200C at 100 hours. At 1375C both systems exhibited creep deformation during testing and low strengths which was characteristic of SSN at that temperature.

The ZrO<sub>2</sub>/sol-gel Al<sub>2</sub>O<sub>3</sub>/SSC and ZrO<sub>2</sub>/laser-textured SSC systems exhibited no significant changes in strength. ZrO<sub>2</sub>/sol-gel Al<sub>2</sub>O<sub>3</sub>/SSN exhibited no change in strength at room temperature and 1375C, and showed higher strength at 1000C which is believed to be coincidental. ZrO<sub>2</sub>/laser-textured SSC exhibited a slight drop in strength at room temperature, but strengths equivalent to as-machined SSC at 1000C and 1375C.

Glass build-up occurred on the ZrO<sub>2</sub> coatings under the hot pressed SiC flexure load pins during testing at 1375C. This glass formation suggested there may be some sticking problems during the 1375C coated contact testing (SSC contact pins are to be used against the ZrO<sub>2</sub> coatings). To assess this problem, studies were conducted to evaluate the sticking resistance of EB-PVD ZrO<sub>2</sub> coatings). To assess this problem, studies were conducted to evaluate the sticking resistance of EB-PVD ZrO<sub>2</sub> against itself and the three substrate materials. The results suggested the ZrO<sub>2</sub> coating sticks to SiO<sub>2</sub> formers (e.g., Si<sub>3</sub>N<sub>4</sub> and SiC) but not to itself. Therefore, in heat engine applications, both contacting surfaces may need to be coated to prevent sticking.

Cyclic Thermal Testing. Each coating/substrate system was subjected to cyclic thermal exposures to 1000C and 1375C to assess spalling resistance. The results were encouraging considering the large thermal expansion mismatch between the coating and the substrate.  $\rm ZrO_2/CVD$   $\rm Al_{2}O_{3}/RBSN$  had the best spalling resistance as no specimens exhibited spalling.  $\rm ZrO_2/laser-textured$  SSN and  $\rm ZrO_2/laser-textured$  SSC performed

nearly as well; minor edge spalling occurred in some cases during 1000C testing.  $ZrO_2/CVD$  AlN/RBSN exhibited relatively good spalling resistance as only one of four spalled during 1375C tests.  $ZrO_2/sol$ -gel Al $_2O_3/SSN$  and  $ZrO_2/sol$ -gel Al $_2O_3/SSC$  exhibited poor spalling resistance as frequent spalling was observed.

Contact Testing. The contact damage resistance of RBSN, SSN, and SSC was significantly higher than originally thought. As a result, thresholds had to be established using point contact testing, rather than line contact testing. Contact damage was observed using point contact loads as low as 11.3 kg for RBSN and SSC. Contact damage occurred consistently using 22.7 kg point contact loads. SSN exhibited better contact damage resistance than RBSN and SSC. Loads of 27.2 kg were required to achieve conditions for damage. Contact damage was observed consistently using a 39.3 kg point contact load.

Coated specimens were also evaluated for friction characteristics during coated contact testing. Surprisingly, the ZrO<sub>2</sub> coatings had higher friction coefficients than the uncoated substrates and the coefficients of friction increased with temperature. The SSC contact pin may deform the low modulus coating to create a plowing effect, which would increase the apparent friction. Oxidation of the SSC contact pin may have caused the friction increase at elevated temperatures. The apparent room temperature friction may be reduced by depositing a dense layer over the porous coating to reduce the plowing effect; and the friction increase at elevated temperatures may be eliminated by using ZrO<sub>2</sub> coated contact pins so that two stable oxides are in contact.

The coating/substrate systems were more resistant to spalling at 1000C than at room temperature, 1200C, and 1375C. It appears that the residual stresses at the interface may be relieved near the coating deposition temperatures.

# 3.4 TASK 4 - POST-CONTACT STRENGTH TESTING

During Task 4, the specimens contact tested during Task 3 were flexure tested at room temperature to measure the retained strengths relative to the coated strengths generated under Task 3 (see Section 3.3.2.2, Table 18). Fractography was performed on each specimen to determine the nature of failure (i.e., contact versus noncontact failure).

# 3.4.1 Strength Test Procedures

The specimens subjected to contact stress conditions in Task 3 were flexure tested with the coated and contact tested side of the specimen in tension. The tests were conducted at room temperature using the test procedures described in Section 3.2.3. After flexure testing, all fracture surfaces were examined up to 40% magnification using an optical binocular microscope. Selected specimens were analyzed further using SEM and ED% analysis.

Table 18. Average retained strengths of contact tested specimens

	Retained strength (MPa)			
	Temperature (C)	Contact failures	Noncontact failures	
ZrO <sub>2</sub> /CVD Al <sub>2</sub> O <sub>3</sub> /RBSN	RT	(0)*	220	
2, 2, 3,	1000	87 (2)	223	
	1375	181 (1)	230	
ZrO2/CVD Aln/RBSN	RT	(0)	238	
4	1000	110 (2)	265	
	1375	(0)	185	
ZrO2/sol-gel Al2O3/SSN	RT	(0)	370	
2. 2 2 3	1000	(0)	393	
	1200	325 (10)		
ZrO2/laser-textured SSN	RT	(0)	415	
•	1000	(0)	424	
	1200	(0)	397	
ZrO2/sol-gel Al2O3/SSC	RT	(0)	291	
2 2 3	1000	180 (3)	288	
	1375	(0)	287	
ZrO <sub>2</sub> /laser-textured SSC	RT	(0)	267	
-	1000	(0)	242	
	1375	(0)	252	

<sup>\*</sup>Numbers in parenthesis indicate the number of specimens exhibiting contact damage.

# 3.4.2 Strength Test Results

The strength test results from contact tested specimens are shown in Table 18. No contact damage resulted from specimens tested at room temperature regardless of how well the coating held up during contact testing. In the cases where the coating chipped off the substrate, the chipping occurred at the onset of sliding or after the point contact had traversed a significant distance across the test specimen. Therefore, the contact pin was actually in contact with the bare substrate during dynamic (sliding) portion of the contact test exposure. In these cases, contact damage may have been prevented by avoiding the more severe static situations (typically, the static coefficients of friction were much higher). Where the coating chipped at the onset of sliding, the stress buildup during the static portion of the contact test where

relieved when the coating failed. In such cases, the friction coefficients actually increased after chipping. The increase was possibly caused by asperity effects produced by debris from the coating increased the apparent friction. It is possible that the presence of debris provided contact damage resistance.

For specimens tested at 1000C, the coatings prevented contact stress damage in most cases, but some contact stress failures were observed. For ZrO<sub>2</sub>/CVD Al<sub>2</sub>O<sub>3</sub>/RBSN two of ten specimens exhibited contact damage. The strength loss was to 87.2 MPa, a 59 percent drop. Surprisingly, these were specimens which did not exhibit chipping of the coating during contact testing. Two specimens from the ZrO<sub>2</sub>/CVD AlN/RBSN system also exhibited contact damage during 1000C contact testing. Neither of these specimens exhibited spalling during contact testing (all specimens from this system survived 1000C contact exposures). The strength loss was to 110 MPa, a 53 percent drop in strenth. Three specimens from the ZrO<sub>2</sub>/sol-gel Al<sub>2</sub>O<sub>3</sub>/SSC system exhibited contact damage during 1000C contact testing. The strength loss was to 180 MPa, a 36 percent strength loss. In all cases, the coating remained adherent during contact testing.

At 1200C all specimens from the  $\rm ZrO_2/sol-gel~Al_2O_3/SSN$  system exhibited contact damage. As noted in Section 3.3.5.2, all  $\rm ZrO_2/sol-gel~Al_2O_3/SSN$  specimens exhibited spalling during contact testing at 1200C. The average strength loss was 15 percent relative the baseline, but strength losses as high as 50 percent were observed. No  $\rm ZrO_2/laser-textured~SSN$  specimens exhibited contact stress damage. All failure origins were the surface cavities produced by the laser-texturing pretreatment.

At 1375C no contact damage was exhibited for  $ZrO_2/CVD$  AlN/RBSN,  $ZrO_2/sol-gel$  Al $_2O_3/SSC$ , and  $ZrO_2/laser-textured$  SSC. Only one specimen exhibited contact damage for the  $ZrO_2/CVD$  Al $_2O_3/RBSN$  system. The resulting strength loss was 14 percent.  $ZrO_2/CVD$  AlN/RBSN did exhibit some strength loss relative to baseline which did not result from contact damage. Speculation is that the 1375C exposure may have caused some degradation of the coating substrate interface.

# 3.4.3 Fractography

SEM and EDX analysis was performed on selected contact and strength tested specimens to further characterize the failures.

As mentioned in the previous section, a few contact damage failures did occur even though the coating remained adherent during contact testing. This damage was observed for 1000C contact testing only. Figure 51 illustrates a contact flaw approximately 250 microns deep. This deep crack was the most severe observed. In most cases, the crack was 25 to 75 microns deep. In all cases the failure origin was under the initial contact point (i.e., where sliding was initiated). Evidently under certain conditioning, well adhered coatings can transmit damage to the substrate.

Since the coatings were more adherent at 1000C (see Section 3.3.5.2), the specimens which did not exhibit contact stress damage may

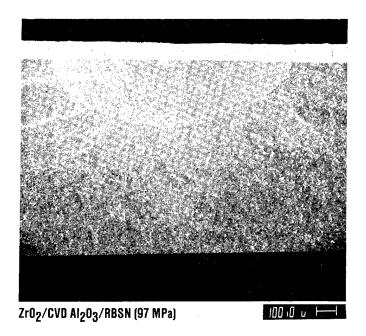


Figure 51. Contact damage occurred in a few cases for which the ZrO<sub>2</sub> coating remained adhered throughout contact testing.

have cracked at the interface or within the coating to relieve the contact stresses generated. At 1375C chipping occurred in all cases, and no contact damage occurred.

When coating chipping occurred, the only contact damage observed was at 1200C for the ZrO2/sol-gel Al2O3/SSN system. All ten specimens contact tested at 1200C exhibited contact damage. The damage for one specimen appeared to be more of a slow crack growth effect due to softening of the grain boundary phases in the substrate. Figure 52 shows a large, wide crack located under contact test initiation point which resulted in a 50 percent strength loss. This crack evidently grew during the 30 minute normal loading of the 45.4 kg point contact and/or the point contact displacement. The contact damage observed for the other specimens appeared to be purely mechanical and resulted in a lesser strength loss averaging approximately 10 percent. ZrO2/laser-textured SSN specimens were tested at 1200C under the same conditions, (i.e., 45.4 kg point contact loading) but did not exhibit any contact damage.

### 3.4.4 Task 4 Summary and Conclusions

The coatings successfully prevented contact damage during room temperature contact testing. However, the coatings were damaged during testing in most cases and, therefore, are not expected to provide contact damage resistance under cyclic loading conditions.

The ZrO<sub>2</sub> coatings exhibited better adherence at 1000C, spalling in only a few instances. Contact damage was prevented in most cases, but a few specimens did exhibit contact damage even though the coating remained adherent throughout testing. It appears the EB-PVD coating's



ZrO2/SOL-GEL AI203/SSN

Figure 52. A large, wide crack in the SSN substrate resulting from from contact stress damage caused a 50 percent strength loss.

columnar microstructure can transmit damage to the substrate under certain conditioning. The weakly adhered coatings appeared to protect the substrate from contact damage sacrificially.

During 1200C and 1375C contact testing, contact damage was prevented for all systems except the  $\rm ZrO_2/sol-gel~Al_2O_3/SSN$  system. All specimens from this system spalled during 1200C and exhibited contact damage. The undamaged systems also exhibited frequent spalling, although the  $\rm ZrO_2/laser-textured$  SSN had a few survivors at 1200C.

### 4.0 CONCLUSIONS

Conclusions related to the application and adherence, and to the strength, friction, and contact stress damage resistance are summarized below:

# Application and Adherence

- The EB-PVD process can deposit controlled coatings of stabilized ZrO<sub>2</sub> with good to excellent adherence on 320 grit diamond ground surfaces of RBSN, SSN, and SSC. STEM and EDX results suggest that the bond interface is a zirconium silicate.
- 2. Mechanical adherence contributes significantly to the overall adherence of EB-PVD ZrO<sub>2</sub> coatings. RBSN, which has the roughest surface of the substrate materials due to its 15 percent residual porosity, consistently exhibited better adherence than the dense SSN and SSC substrates. Also, lapping the 320 grit ground surfaces prior to coating resulted in poorer adherence in most cases.
- 3. Roughening the substrate surface prior to applying the EB-PVD ZrO<sub>2</sub> coating improves the mechanical adherence. Laser-texturing of SSC and SSN substrates signifi-cantly improved the coating adherence both before and after oxidation without the use of an oxygen diffusion barrier.
- 4. Oxidizing EB-PVD ZrO<sub>2</sub> coated RBSN, SSN, and SSC degrades the level of coating adherence significantly. The adherence degradation is due to the growth of the silicate layer at the interface. The high oxygen permeability of ZrO<sub>2</sub> allows oxygen transport to the coating/substrate which enables silicate growth at high temperatures.
- 5. Depositing an oxygen diffusion barrier between the EB-PVD  $\rm ZrO_2$  coating and the  $\rm Si_3N_4$  or SiC substrate is a feasible approach for reducing or eliminating adherence degradation caused by interfacial oxidation.
- 6. CVD Al<sub>2</sub>O<sub>3</sub> and CVD AlN interlayers work well for the porous RBSN substrates but not for the dense SSN and SSC substrates. Apparently, thermal expansion mismatch between the dense interlayer and the substrate result in interfacial cracking, but the residual porosity in the RBSN enables islands of interlayer to remain adherent due to rooting into the substrate pores.
- 7. Sol-gel Al<sub>2</sub>O<sub>3</sub> interlayers are preferred for SSN and SSC substrates over CVD Al<sub>2</sub>O<sub>3</sub> since relatively good adherence was retained after oxidation exposure. The 50 percent porous solgel interlayer (submicron sized porosity) evidently provides improved strain tolerance which limits interfacial cracking due to thermal expansion mismatch, but, at the cost of the cohesive strength of the interlayer.

# Strength, Friction, and Contact Damage

8. RBSN, SSN, and SSC exhibit significantly higher contact damage resistance than originally assumed.

EB-PVD Y<sub>2</sub>O<sub>3</sub> stabilized ZrO<sub>2</sub> coatings increase the contact 9. damage resistance of Si3N4 and SiC ceramics. However, the contact damage resistance provided appears to be sacrificial protection.

EB-PVD Y2O3 stabilized ZrO2 coatings exhibit significantly 10. higher coefficients of friction during contact testing than dense Y2O3 stabilized and transformation toughened ZrO2. The high compliancy of the porous EB-PVD coating may allow the contact pin to imbed in the coating to create a plowing effect

which increases the apparent friction.

 ${\tt EB-PVD}$   ${\tt Y}_2{\tt O}_3$  stabilized  ${\tt ZrO}_2$  exhibits an increasing coefficient 11. of friction with temperature when tested against SiC. This increase is probably due to the formation of SiO2. ZrO2 tested against itself, may not produce this increased friction. Compatibility studies at 1375C showed that a ZrO2 coated specimen will stick severely to Si3N4 and SiC but will not stick to another ZrO2 coated specimen. A suggestion for engine application of ZrO2 coatings is that both mating surfaces of Si3N4 and/or SiC components should be coated to prevent sticking due to oxide formation.

Well bonded EB-PVD ZrO2 coating can transmit damage to the 12. substrate under certain conditions. Further work is needed to better understand this phenomenon and to prevent the damage

transmittance.

Dense, strong interlayers may result in residual stress 13. effects due to thermal expansion mismatch. During this program, CVD Al<sub>2</sub>O<sub>3</sub> and CVD AlN interlayers reduced the room temperature strength of the substrate, but the strength recovered at elevated temperature. In future studies more emphasis should be placed on minimizing thermal expansion mismatches in order to minimize residual stress effects.

Laser-texturing can reduce the room temperature strength of -14. Si3N4 and SiC substrates, although the strength is recovered at elevated temperatures. Microcracks extending from the laser produced cavities limit substrate strength. research is required on laser-texturing of Si3N4 and SiC to

minimize the resulting strength losses.

### REFERENCES

- 1. Advanced Gas Turbine (AGT) Powertrain System Development for Automotive Applications, second Semiannual Progress Report, Contract No. CR-165329, GTEC Report No. 31-3725(2), July 1981.
- 2. D.W. Richerson and K.M. Johansen, <u>Ceramic Gas Turbine Engine Demonstration Program</u>, <u>Final Report</u>, <u>DARPA Contract N00024-76-C5352</u>, May 1982, GTEC Report 21-4410.
- 3. J.M. Criscione, S. Sarian, H.F. Volk, et al., <u>High Temperature Protective Coatings for Graphite</u>, <u>Summary Technical Report for Work Conducted 1 June 1963 to 31 May 1964</u>, from ML-TDR-64-173, Part II, 1964.
- 4. Personal communication with Dr. A.G. Evans of the University of California, Berkeley, and Daniel Lee of Temescal, Berkeley, CA, during program coordination meeting held September 12, 1984, in Berkeley, CA.
- 5. <u>Highly Reliable High Performance Thermal Barrier Coatings</u> (TBC), Technical Proposal, GTEC Report 21-5132-2, August 23, 1984.
- 6. Unreported research conducted jointly by GTEC and Special Processes of Arizona in 1980.
- 7. D.W. Richerson, W.D. Carruthers, L.J. Lindberg, "Contact Stress and Coefficient of Friction Effects on Ceramic Interfaces," Surfaces and Interfaces in Ceramic and Ceramic Metal Systems, Materials Science Research, Volume 14, pp. 661-676. Edited by Joseph Pask and Anthorny Evans, Plenum Press, NY 1981.
- 8. J.R. Smyth, D.W. Richerson, "High Temperature Dynamic-Contact Behavior of Sintered Alpha Silicon Carbide," <u>Ceramic Engineering and Science Proceedings</u>, Vol. 4, No. 7-8, 1983, pp. 663-673.
- 9. L.J. Lindberg and D.W. Richerson, "Comparison of the Contact Stress and Friction Behavior of SiC and ZrO<sub>2</sub> Materials," <u>Ceramic Engineering and Science Proceedings</u>, Vol. 6, No. 7-8, 1985, pp. 1059-1066.
- 10. L.J. Lindberg, <u>Contact Stress and Frictional Behavior of</u>
  <u>Three Commercially Available Transformation Toughened Zirconia Materials</u>, <u>Masters Thesis</u>, <u>Arizona State University</u>, 1984.

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